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Strip Theory Method of Calculation for Airscrews on High-Speed Aeroplanes

By

C. N. H. Lock, M.A., F.R.Ae.S., R. C. PANKHURST, Ph.D., A.R.C.S.,
AND J. F. C. Conn, D.Sc., M.I.N.A.,
of the Aerodynamics Division, N.P.L.

With an
APPENDIX

By

J. N. Veasey, B.A.,
of the Aerodynamics Division, N.P.L.

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C. N. H. LOCK, M.A., F.R.Ae.S., R. C. PANKHURST, Ph.D., A.R.C.S., and
J. F. C. CONN, D.Sc., M.I.N.A.,

With an Appendix by J. N. Veasey, B.A.,
of the Aerodynamics Division, N.P.L.

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* This report contains material from the following unpublished papers : A.R.C. Reports 5078 (April, 1941), 5233 (July, 1941), 5939 (July, 1942), 6512 (February, 1943) and 8962 (September, 1945).

Summary.—The report describes a simplified 8-point strip theory method of calculating the free-air performance of a propeller up to tip Mach numbers near the velocity of sound. It is based on the assumptions of R. & M. 1674⁴ and 1849⁵ together with the further simplifying assumption that the (C_L, α) curve is straight (valid below the stall) and that the (sC_L, φ) curve also is straight (valid for $J > 1.0$). The report includes tables of parameters (b, τ, ζ and q) which are required in the calculations as functions of J, r_c and N for eight standard radii ($r_c = 0.3, 0.45, 0.6, 0.7, 0.8, 0.9, 0.95, 0.975$) for the range of values of J from 1.0 to 7.0; these are of universal application. In addition, tables of section data (C_L and C_D as functions of M and α) for various section shapes are required; these are given for Clark Y sections over a range of thickness in R. & M. 2036⁷; they were derived, by methods described in R. & M. 2020⁸, from overall measurements of thrust and torque on full scale propellers at low values of J in the Royal Aircraft Establishment 24-ft. tunnel and are subject to revision in the light of subsequent experimental research.

The method determines values of torque grading (q_c), thrust grading (t_c) and power loss gradings (p_{c1} and p_{c2}); these are integrated arithmetically to give free air thrust, torque and efficiency (k_T, k_Q and η) by means of integrating coefficients which are also tabulated. A method is described of determining the contribution of the portion of the blades inside the radius $r_c = 0.3$.

LIST OF SYMBOLS

a	Slope of curve of α against sC_L (assumed linear).
a_0	Value of a at low speed.
a_h	Speed of sound at height h .
A_0	Slope of low-speed lift curve ($dC_{L,0}/da$).
b	Slope of curve of φ against sC_L (assumed linear).
B_0	Factor giving variation of C_{D0} with α_0 .
c	Chord of blade element at radius r .
C_0	Value of C_{D0} when $\alpha_0 = 0$.
C_1, C_2, \dots	Integrating coefficients.
C_D	Drag coefficient of blade element.
C_{D0}	Low-speed drag coefficient.
C_{DS}	Compressibility-increase in C_D over its low-speed value.
C_L	Lift coefficient of blade element.
C_{L0}	Low-speed lift coefficient.
C_{LS}	Increase in C_L over its value at the lift critical speed.
D	Airscrew diameter.
h	Operating altitude of aircraft.
J	Advance ratio (V/nD).
k_P	Total power loss coefficient (Power wastage/ $2\pi\rho n^3 D^5$).
k_{P0}	Low-speed component of the profile drag power loss coefficient.
k_{P1}	Induced power loss coefficient.
Δk_{P1}	Blade root power loss coefficient.
k_{PS}	Compressibility component of the profile drag power loss coefficient.
k_Q	Torque coefficient ($Q/\rho n^2 D^5$).
k_T	Thrust coefficient ($T/\rho n^2 D^4$).
M	Mach number of blade element.
M_D	Critical Mach number for drag.
M_L	Critical Mach number for lift.
n	Rotational speed (r.p.s.).
N	Number of blades.
p_c	Grading coefficient of the total power loss: $dk_P/d(r_c^2)$.
p_{c0}	Grading coefficient of the low-speed component of the profile drag power loss.

List of Symbols—*contd.*

\dot{p}_{c1}	Grading coefficient of the induced loss.
\dot{p}_{c2}	Grading coefficient of the profile drag power loss.
\dot{p}_{cs}	Grading coefficient of the compressibility component of the profile drag power loss.
q	Factor giving the power loss grading coefficient : $(\pi^3/16)r_c^3 \sec^3\phi_0$.
q_c	Torque grading coefficient : $dk_q/d(r_c^2)$.
Q	Torque.
r	Radius at blade element.
r_c	Fractional radius at blade element (r/R).
R	Tip radius.
s	Solidity ($Nc/2\pi r$).
t	Thickness of blade section.
t_c	Thrust grading coefficient : $dk_T/d(r_c^2)$.
T	Thrust.
V	Forward speed.
w_1	Inflow velocity.
W	Resultant relative air velocity at blade element.
W_0	Geometrical velocity of blade element.
α	Incidence of blade element.
α_0	Incidence referred to zero-lift datum.
β	Inflow angle ($\varphi - \varphi_0$).
Γ	Circulation taken over blade element.
ϵ	Zero-lift angle.
ϵ_0	Zero-lift angle at low speed.
ζ	Factor giving the torque grading coefficient : $(\pi^3/16) r_c^3 \sec^2\varphi_0$.
η	Airscrew efficiency.
$\Delta\eta$	Efficiency loss due to blade roots.
θ	Blade angle.
κ	Inflow factor corresponding to helicoid angle φ .
κ_0	Inflow factor corresponding to helicoid angle φ_0 .
ρ	Air density.
τ	Factor giving the thrust grading coefficient : $(\pi^3/8)r_c^2 \sec^2\varphi_0$.
φ	Angle between plane of rotation and relative air velocity at blade element.
φ_0	Angle between plane of rotation and geometrical velocity of blade element.
Ω	Rotational speed (radians per second).

1. *Introduction.*—The method of airscrew strip theory calculation described in the present report has been gradually developed in serial form¹⁻⁵. In preparing for printing the part of the work so far unpublished, it therefore seemed worth while to recapitulate the various basic assumptions in some detail by way of introduction.

The first and most fundamental assumption is that the resultant force on an elementary strip of the airscrew blade (of length dr at radius r), is the same as it would be if the element formed part of a two-dimensional aerofoil, situated in an airstream the magnitude and direction of whose velocity is to be determined. This velocity is the resultant of the geometrical velocity of the blade element and a certain "interference" velocity due to the trailing vortex system.

Thus the problem before us resolves itself into two distinct parts—the method of dealing with the two-dimensional force coefficients and the determination of the “interference velocity”.

The recent developments with which the present report is concerned deal mainly with the first problem; only minor changes directed to increased simplicity in computation have been made in the methods of solution of the second problem.

We shall treat the second problem in some detail, but before doing so a few remarks on the first problem are desirable.

Recent developments have been chiefly concerned with the effect of high “Mach number” on the two-dimensional section lift and drag coefficients. The ultimate aim should be to determine these coefficients from actual experiments on two-dimensional aerofoils with the addition, if necessary, of empirical corrections for any failure of the theoretical calculation of the interference velocity. The force coefficients referred to in the present paper were obtained (R. & M. 20367) mainly by “back-figuring” from tests of actual propellers and would, therefore, include these empirical corrections. The data are essentially provisional, applying to a particular form of blade section, and being subject to revision whenever further experimental results become available.

2. Determination of the Interference Velocity.—The method is based on the Prandtl-Betz assumption of replacing the airscrew blade by a lifting line consisting of a “bound vortex” whose strength at any radius is equal to the circulation round the blade at that radius. The strength of the vortex sheet springing from the blade adjusts itself so as to satisfy the condition of continuity of vorticity; for an airscrew it is of spiral form. The “interference velocity” at the blade element is to be calculated on the assumption that the flow is everywhere irrotational outside this vortex system. The resultant force on a bound vortex is normal to the resultant relative velocity W ; accordingly, the drag is neglected in calculating the interference velocity.

Figure 1 shows a cross-section of the air flow relative to the blade element at radius r . According to the general theorem that the vortex lines coincide with the relative streamlines, the direction of the trailing vortex at radius r , at the point where it leaves the blade, coincides with the direction of W , and the initial angle of pitch of the helical trailing vortex is equal to ϕ . It is also worth noting here that the theory for determining interference velocity is essentially a first order theory, in that the interference velocity ratio w_1/W is treated as a small quantity of which the square is to be neglected. The difference between the assumptions that the angle of pitch of the trailing vortex is equal to φ or to φ_0 is in fact of the second order, but it is worth while to prefer the former assumption in the initial stages because it leads to results which agree exactly, for the limiting case of an infinite number of blades, with results deducible from consideration of linear and angular momentum and energy.

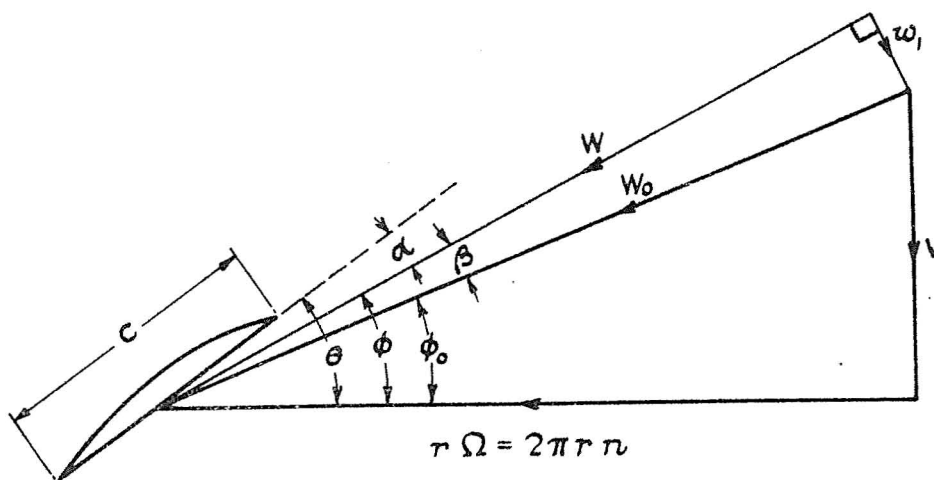


FIG. 1

It may be shown in a general manner that w_1 is normal to the vortex line and, therefore, to W as shown in Fig. 1. The essentially difficult part of the problem is to determine a relation between the magnitude of w_1 and the strength and configuration of the trailing vortices. It has so far been found necessary to ignore the variation of pitch and radius of the trailing vortices and to treat them as regular helices. It is then possible to use the standard artifice of the Prandtl theory of the lifting wing to show that the actual value of w_1 in the presence of the semi-infinite trailing vortices is one half the value that would occur for similar vortices of infinite length.

An expression for the velocity field of a single infinite helical vortex has been given by Lamb⁶ in terms of Bessel functions, but the difficulty of the numerical evaluation of the principal value of an infinite integral along the blade, together with subsequent successive approximation to fit the solution to a particular working condition of a particular airscrew, makes the use of this solution prohibitive.

The method to be used here is based on Goldstein's solution for a particular case. Considering a helical vortex sheet extending to infinity in both directions, $2w_1$ is equal to the component velocity of the vortex sheet normal to itself and $2w_1 \sec \varphi$ is equal to the velocity parallel to the axis of the airscrew. For the particular case in which this quantity is independent of the radius, the trailing vortex sheet may be considered as a rigid surface moving parallel to the axis with velocity $2w_1 \sec \varphi$. This case was proved by Betz to give minimum induced energy loss.

Goldstein's solution may be considered as giving (in tabular form) the equivalent vortex strength, at any radius r , of a series of rigid infinite helicoidal surfaces (one for each blade of the airscrew) of radius R and angle of pitch φ , moving parallel to the axis with velocity $2w_1 \sec \varphi$. The circulation Γ round the blade at radius r can be immediately determined as the total strength of the vortex sheet outside that radius.

Since Γ is obviously proportional to w_1 and of the dimensions of $w_1 r$, the tables could be expressed in non-dimensional form by tabulating $\Gamma/w_1 r$ as a function of $r_c (= r/R)$ and φ . Actually it is convenient to use the equivalent coefficient κ defined by the equation

$$\kappa = \frac{\Gamma}{w_1 r} \cdot \frac{N}{4\pi \sin \varphi} \quad \dots \quad (1)$$

The values of κ are tabulated as functions of $\sin \phi$, r_c and N in Table 1a, whose derivation is discussed in Appendix 1 and Appendix 2.

The coefficient κ has so far been defined as determining the value of the circulation at any radius, for the particular case where the trailing vortex sheets are rigid surfaces moving with a given velocity. The artifice by which the results may be applied in the general case consists in the assumption that the relation connecting κ with r_c , φ and N , is *independent of the conditions at other radii*. The significance of this assumption may be made clearer by reference to Figure 2.

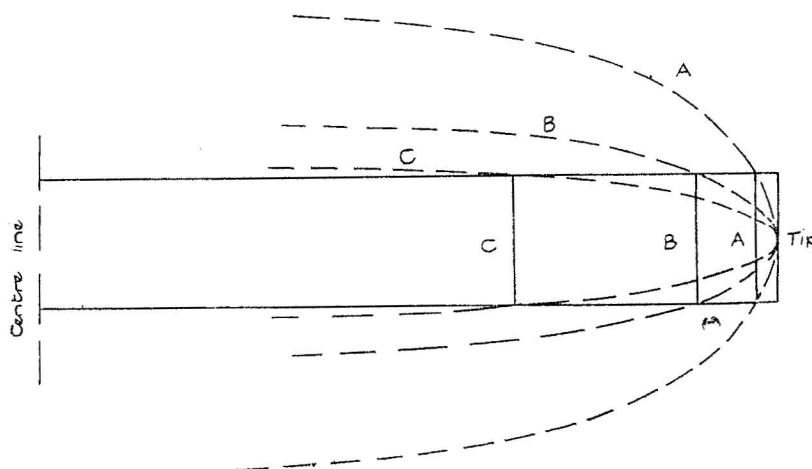


FIG. 2.

Consider the section B of a square-tipped blade and suppose that the dotted curve B represents the plan form of a blade which, with the same pitch distribution, would give rise to a wake of Betz' form for the given working condition. The proposed method calculates the interference velocity for the section at radius B on the assumption that it forms part of a blade having the dotted plan form B . For other radii the equivalent plan form is different. Thus the blade elements are treated as being effectively independent, although the value of the coefficient κ depends on r_c .

The following further points are worth noting.

The coefficient κ is defined in equation (1) in such a way that its value tends to unity as the number of blades tends to infinity. The solution for an infinite number of blades can be obtained by more elementary methods, and in this limiting case the blade elements are independent.

The coefficient κ tends to zero at the blade tip for any finite number of blades. Equation (1) then ensures that, even for a square-tipped blade, the circulation and lift coefficient tend to zero at the blade tip. This requires, according to Fig. 2, an "equivalent plan form" of infinite solidity for a square-tipped blade.

The justification of the proposed use of equation (1) is further discussed in R. & M. 1521³. This equation will accordingly be used as the basis of the calculation of interference velocity throughout the present report.

3. *Forces on the Blade Element: Grading Coefficients.*—The fundamental relation of equation (1) is next expressed in a form more convenient for practical use by introducing the lift coefficient at the blade element. If c denotes the chord, the lift on the element may be written—

$$\frac{1}{2}C_L \rho c \delta r W^2 = \delta L = \rho W \Gamma \delta r \quad \dots \dots \dots (2)$$

Hence
$$2\Gamma = cWC_L \quad \dots \dots \dots (3)$$

Substitution for Γ in (1) then gives at once

$$\frac{Nc}{2\pi r} C_L = 4\kappa \frac{w_1}{W} \sin \varphi \quad \dots \dots \dots (4)$$

Reference to the velocity diagram (Fig. 1) shows that $w_1/W = \tan \beta$, so that

$$sC_L = 4\kappa \tan \beta \sin \varphi \quad \dots \dots \dots (5)$$

where $s (= Nc/2\pi r)$ denotes the blade solidity at the radius considered.

Also from the velocity diagram we have—

$$\theta = \alpha + \varphi \quad \dots \dots \dots (6)$$

and

$$\varphi = \varphi_0 + \beta \quad \dots \dots \dots (7)$$

Assuming a known lift curve for the section, we thus have, with equations (5), (6) and (7), four relations which serve to determine four unknowns. Usually the blade angle θ is known and the equations are to be solved for C_L , α , φ and β .

At high rates of advance a simplification is possible, due to the fact that sC_L is very nearly linear with φ . In this case equation (5) may be written, to the first order in β , as—

$$sC_L = 4\kappa_0 \beta \sin \varphi_0 \quad \dots \dots \dots (8)$$

or
$$\beta = bsC_L \quad \dots \dots \dots (9)$$

with
$$b = 1/4 \kappa_0 \sin \varphi_0 \quad \dots \dots \dots (10)^*$$

so that the three equations (6, 7 and 9) are all linear in the unknowns.

* These formulae are believed to be sufficiently accurate for all values of J greater than 2.0 and should give fairly accurate results down to $J = 1.0$.

4. *Integrating Coefficients.*—It is finally necessary to integrate the values of t_c , q_c , p_{c1} and p_{c2} with respect to r_c^2 between the limits 0.09 and 1.0* to give values of k_T , k_Q , k_{P1} and k_{P2} . This may be done graphically, but more consistent results may be obtained with less labour by the use of integrating coefficients C_n (given in Table 3a taken from R. & M. 2043⁹) in formulae such as—

$$k_Q = C_1 q_{c1} + C_2 q_{c2} + \dots + C_8 q_{c8} \quad \dots \quad (35)$$

where the suffices refer to the eight standard values of r_c . If separate values of k_{P1} and k_{P2} were not required, it would be sufficient to evaluate

$$k_P = k_{P1} + k_{P2} \quad \dots \quad (36)$$

by integrating

$$p_C = p_{c1} + p_{c2} \quad \dots \quad (37)$$

Finally, the efficiency η may be obtained by using equation (34).

5. *Calculation of Mach Number.*—The Mach number of a blade section is taken to be—

$$M = W_0/a_h \quad \dots \quad (38)$$

where

$$W_0 = \Omega r \sec \varphi_0 \quad \dots \quad (39)$$

and a_h is the speed of sound at the height h at which the aircraft is flying.†

Hence

$$M = (\Omega R/a_h) r_c \sec \varphi_0 \quad \dots \quad (40)$$

where

$$J = V/nD \quad \dots \quad (41)$$

and

$$\tan \varphi_0 = J/\pi r_c \quad \dots \quad (42)$$

Values of φ_0 and $r_c \sec \varphi_0$ are tabulated against J for standard values of r_c in Tables 2a and 2b.

6. *Formulae for (C_L, α) in Terms of Thickness Ratio and Mach Number.*—The principal object for which the method is designed is the calculation of the performance of airscrews for high-speed aeroplanes on the basis of the section lift and drag data reported in R. & M. 2036⁷. Over the range of incidence likely to be used at high speed it is sufficiently accurate to assume a linear lift curve for a given Mach number.

The lift coefficient of the section of an airscrew blade is a function of incidence, Mach number and thickness-chord ratio. It is conveniently expressed in terms of its low-speed value C_{L0} , which is assumed to be a linear function of incidence α over the range $0 < C_{L0} < 0.6$ according to the equation

$$C_{L0} = A_0 \alpha_0 \quad \dots \quad (43)$$

where $\alpha_0 (= \alpha + \varepsilon_0)$ denotes the incidence referred to the zero-lift datum. The low-speed slope A_0 is taken as 0.1 per degree for values of t/c less than 17 per cent. for airscrew sections of shape similar to Clark Y. For larger values of t/c its value is taken provisionally from Table 1 of R. & M. 2036⁷; this table was based on experiments in the compressed air tunnel¹¹ on aerofoils of thickness-chord ratio up to 30 per cent.

The low-speed zero-lift angle ε_0 is taken to be that given by the theory of thin aerofoils¹², using the rapid method of evaluation given in R. & M. 1914¹³.

* The lower limit of 0.09 for r_c^2 (corresponding to 0.3 for r_c) was chosen arbitrarily; it is necessary to estimate the contribution of the portion of the blade inside this radius (blade root losses) by other methods (§ 8).

† The value of a_h as a function of height under standard conditions is given in R. & M. 1891¹⁰.

The lift coefficient C_L , considered as a function of Mach number as well as α , is defined by distinct formulae in two separate ranges in which the values of C_L , a , ε are here distinguished by the suffices 1 and 2 respectively, the boundary between the two ranges corresponding to a critical Mach number M_L which for values of α_0 less than 3 deg. is an empirical function of thickness-chord ratio only, but depends also on incidence at higher values of α_0 . Values of M_L are given in Table 3 of R. & M. 2036⁷.

Range 1 includes all values of M less than M_L . In this range the lift coefficient rises with Mach number at a rate which for sections thinner than 16 per cent. follows Glauert's equation—

$$C_{L1} = (1 - M^2)^{-1/2} C_{L0} \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (44)$$

Comparison with equation (11) then gives the following of a and ε :—

$$a_1 = (1 - M^2)^{1/2} a_0 \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (45)$$

$$\varepsilon_1 = \varepsilon_0 \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (46)$$

where

$$a_0 = 1/sA_0. \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (47)$$

Values of $(1 - M^2)^{1/2}$ and $(1 - M^2)^{-1/2}$ are given in Table 2 of R. & M. 2036⁷.

For thicker sections the rate of rise with Mach number becomes progressively less than $(1 - M^2)^{-1/2}$; use is then made of Table 5 of R. & M. 2036⁷.

Range 2 includes all values of M greater than M_L . In this range C_L is given by the equation—

$$C_{L2} = (1 - M_L^2)^{-1/2} C_{L0} + C_{LS} \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (48)$$

where C_{LS} is an empirical function of $(M - M_L)$ given in Table 4 of R. & M. 2036⁷. Comparison with equation (11) now gives the following values of a and ε :—

$$a_2 = (1 - M_L^2)^{1/2} a_0 \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (49)$$

$$\varepsilon_2 = \varepsilon_0 + (1 - M_L^2)^{1/2} C_{LS}/A_0. \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (50)$$

Substitution of the appropriate values of a and ε in equations (13), (9), (7) and (11) will then determine the appropriate values of C_L , φ and α for both ranges, except that when α_0 exceeds 3 degs. the solution involves a small amount of successive approximation since M_L then depends on α as well as t/c . The principle of the successive approximation (R. & M. 2036⁷) consists in adopting the low-incidence value (M'_L) of M_L (independent of α) and the corresponding value (a') of a as a first approximation, and evaluating $\alpha_0' = a'(\theta - \varphi_0 + \varepsilon_0)/(a' + b)$. If this exceeds 3 degs. a second approximation to M_L is necessary. In practice it is found that this second approximation suffices.

Formulae for C_D in Terms of Mach Number and C_{L0} .—In considering the effect of high Mach number it is convenient, except at high incidence and for thick sections, to write—

$$C_D = C_{D0} + C_{DS} \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (51)$$

where C_{DS} is zero for low values of M .

C_{D0} .—It is assumed that the value of C_{D0} at any radius is given by the equation—

$$C_{D0} = B_0 C_0 \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad \dots \quad (52)$$

where B_0 is a function of α_0 only and C_0 is a function of t/c only, given in Tables 6 and 7 of R. & M. 2036⁷.

C_{DS} .—The compressibility drag C_{DS} is an empirical function of $(M - M_D)$, given in Table 9 of R. & M. 2036⁷, where M_D is a critical Mach number for drag, assumed to be an empirical function of the two variables α_0 and t/c (Table 8 of R. & M. 2036⁷). When dealing with a series of calculations for all of which t/c is the same at a given radius, it is convenient to draw a family of curves of M_D against α_0 , one for each standard radius.

At incidences above 8 deg. for sections thinner than 9 per cent. and above 5 deg. for sections from 9 per cent. to 16 per cent., it is more convenient to use curves of C_D plotted against M directly. These are given as Figs. 1-15 of R. & M. 2036⁷.

It should be emphasised that the above formulae for C_L and C_D represent the best available evidence at the time of writing (August, 1945). If it should prove necessary to modify them when further experimental evidence becomes available, it will still be possible to use the method of § 3 with Tables 1 to 3, provided that a linear relation between C_L and α for given M still holds over a sufficient range.

7. *Specimen Calculation.*—A specimen calculation for one blade section is given in Table 4. For a complete airscrew there would be eight columns corresponding to the eight standard radii.

Values of the following quantities are supposed given—for the whole airscrew: N (number of blades), V (forward speed), n (airscrew revolutions per second), h (operating altitude, to determine the speed of sound a_h), from which are derived $\Omega R/a_h$ and J ; and at each standard radius (r_c): t/c , section shape (determining ε_0), s (solidity) and θ (blade angle).

The three columns of Table 4 give examples respectively of Range 1, Range 2 at low incidence, and Range 2 at high incidence. If $M < M_L$ (Range 1), the entry for $(M - M_L)$ should be " < 0 ", and the value of $(1 - M^2)^{1/2}$ used instead of $(1 - M_L^2)^{1/2}$. If $M > M_L$, (Range 2) the value of $(M - M_L)$ is inserted so as to determine C_{LS} , and $(1 - M_L^2)^{1/2}$ is used instead of $(1 - M^2)^{1/2}$.

The final results for a particular working condition and blade setting of a given airscrew will be values of q_c or t_c , p_{c1} , p_{c0} and p_{cs} for the eight standard values of r_c . The integrating coefficients given in Table 3a are then used to integrate the above quantities between the limits $r_c^2 = 0.09$ and 1.0 to give values of k_Q or k_T , k_{P1} , k_{P0} and k_{PS} . The efficiency (η) follows.

8. *Blade Root Loss.*—The lower limit of integration corresponding to the numerical coefficients of Table 3a is $r_c = 0.3$: $(r_c)^2 = 0.09$. The effect of the portion of the blade between this radius and the spinner is conveniently dealt with^{14, 15} as a separate loss of efficiency, in the calculation of which the contribution of the root sections to the torque is in general negligible. Neglecting the induced loss, the power loss coefficient is given from equation (33) in the form—

$$\Delta k_p = \int_{\text{spinner}}^{0.09} q s C_D d(r_c^2), \quad \dots \dots \dots (53)$$

so that the efficiency loss is simply

$$\Delta \eta = \Delta k_p / k_Q \quad \dots \dots \dots (54)$$

to the first order of small quantities.

Since $q = \frac{\pi^3}{16} r_c^3 \sec^3 \varphi_0 \quad \dots \dots \dots (32)$

and $J = \pi r_c \tan \varphi_0, \quad \dots \dots \dots (42)$

it follows that

$$\Delta k_p = \int_{\text{spinner}}^{0.09} s C_D \operatorname{cosec}^3 \varphi_0 d(r_c^2). \quad \dots \dots \dots (55)$$

APPENDIX 2

Calculation of κ near the Blade Roots

By J. N. Veasey, B.A.

Since it was considered that simple extrapolation from the values of Appendix 1 to smaller radii would be unsatisfactory, and that the algebraic method used before for interpolating for other numbers of blades than 2 and 4 might be unsuitable for the function at the small radii where it attains values greater than unity, the following method was adopted. Use was made of the solution for the limiting case of infinite pitch given by Westwater and Goldstein¹⁶.

(a) *For 2 and 4 Blades.*—For the limiting case of very small radii, no great simplification of the general formulae given by Goldstein¹ could be found, but for infinite pitch¹⁶ we have equations (64) and (65) (quoted in Appendix 1); when $r_c \rightarrow 0$, these reduce to:—

$$\text{For 2 blades } \kappa = 1/\pi r_c \quad \dots \dots \dots (72)$$

$$\text{For 4 blades } \kappa = \frac{8}{\pi^2} \log \frac{1}{r_c} \quad \dots \dots \dots (73)$$

As $r_c \rightarrow 0$, the pitch angle of the helicoidal vortex sheets becomes 90 deg., and so the above relations give an approximation to κ when r_c is very small.

The equations suggest that κ becomes large towards the roots of the blades, and from equation (72) it might be expected that for conditions other than infinite pitch, κr_c for two blades would remain finite when r_c approached zero, and that for four blades κr_c would approach zero logarithmically.

Values of κr_c for two and four blades taken from Ref. 1 and Table 7 of R. & M. 1674⁴ were accordingly plotted against r_c for constant pitch ($r_c \tan \varphi = \text{const.}$), and the curves extrapolated to $r_c = 0$ using the curves of equations (72) and (73) as a guide.

From these curves, values of κ at even intervals of r_c were taken and it was found that by plotting $1/\kappa$ against $1/r_c \sec \varphi$ at constant r_c , a fairly linear and uniform set of curves was obtained. The reciprocal of κ was chosen as the ordinate because κ is usually greater than unity at the root sections, becoming very large towards the propeller axis. For abscissa $1/r_c \sec \varphi$ was used since, besides producing a smooth series of curves, it becomes zero for the condition of infinite pitch (values calculated from Ref. 16); moreover, the high advance ratios fall within a small range of the abscissa.

(b) *3, 5, 6 and 8 Blades.*—Since the formulae in Ref. 1 for the inflow factor for these blade numbers are laborious to compute and involve Bessel Functions which have not been tabulated, only a small number of values were actually calculated for 3 and 6 blades; the other values were interpolated graphically by plotting $1/\kappa$ against both $1/r_c \sec \varphi$ (as for 2 and 4 blades) and $1/N$, using as a guide points obtained from the results for 2 and 4 blades and also from the solution for the limiting case of infinite pitch¹⁶.

Since κ represents the ratio of the mean inflow velocity round the circumference of a circle of given radius to the inflow velocity at the vortex sheets at that radius, this ratio (κ) will become equal to unity for an infinite number of blades, for which the vortex sheets are indefinitely close together. The inflow factor κ was therefore taken to be equal to unity for the case of an infinite number of blades, for all radii and advance ratios.

REFERENCES

- | <i>No.</i> | <i>Author.</i> | <i>Title, etc.</i> |
|------------|--|--|
| 1. | S. Goldstein | On the Vortex Theory of Screw Propellers. <i>Proc. Roy. Soc. A</i> , 123 , 440 (1929). |
| 2. | C. N. H. Lock | Application of Goldstein's Airscrew Theory to Design. R. & M. 1377. November, 1930. |
| 3. | C. N. H. Lock | An Application of Prandtl Theory to an Airscrew. R. & M. 1521. August, 1932. |
| 4. | C. N. H. Lock and D. Yeatman .. | Tables for Use in an Improved Method of Airscrew Strip Theory Calculations. R. & M. 1674. October, 1934. |
| 5. | C. N. H. Lock | A Graphical Method of Calculating the Performance of an Airscrew. R. & M. 1849. August, 1938. |
| 6. | H. Lamb | <i>Proc. Camb. Phil. Soc.</i> 21 , 477 (1923). |
| 7. | A. B. Haines and R. J. Monaghan .. | Section High-Speed Lift and Drag Data for Propeller Performance Calculations. R. & M. 2036. November, 1945. |
| 8. | R. C. Pankhurst and A. B. Haines .. | Account of the Derivation of High-Speed Lift and Drag Data for Propeller Blade Sections. R. & M. 2020. August, 1945. |
| 9. | C. N. H. Lock and A. E. Knowler .. | Integrating Coefficients for Airscrew Analysis. R. & M. 2043. July, 1941. |
| 10. | R. C. Pankhurst and J. F. C. Conn .. | Physical Properties of the Standard Atmosphere. R. & M. 1891. January, 1941. |
| 11. | D. H. Williams, A. H. Bell and W. G. Raymer. | Tests on a 20 per cent. Piercy Aerofoil and on N.A.C.A. 0030 with and without Flaps, in the Compressed Air Tunnel. A.R.C. No. 4511, April, 1940. (To be published.) |
| 12. | H. Glauert | Aerofoil and Airscrew Theory, p. 91. Cambridge University Press, 1926. |
| 13. | R. C. Pankhurst | A Method for the Rapid Evaluation of Glauert's Expressions for the Angle of Zero Lift and the Moment at Zero Lift. R. & M. 1914. March, 1944. |
| 14. | P. A. Hufton | The Calculation of Airscrew Efficiencies at High Speeds. R.A.E. Performance Note No. 18. July, 1940. (Unpublished.) |
| 15. | R. J. Monaghan | Estimation of the Compressibility Losses on Propeller Blade Roots. A.R.C. No. 8371. December, 1944. (Unpublished.) (The data of this report are included in Ref. 7 above.) |
| 16. | S. Goldstein | On the Limiting Value for Infinite Pitch of a Parameter occurring in Airscrew Theory. <i>Proc. Camb. Phil. Soc.</i> , 40 , 146. (1944.) |
| 17. | E. T. Whittaker and G. Robinson .. | The Calculus of Observations, Chapter XI. Blackie, 1940. |

TABLE 1a

Values of $\kappa : r_c = 0.2$

N Sin ϕ	2	3	4	5	6
0.68	1.000 9	1.006 9	1.011 3	1.008 2	1.004 3
0.70	1.009 11	1.015 11	1.014 5	1.010 3	1.007 3
.72	.020 12	.026 12	.019 7	.013 5	.010 5
.74	.032 13	.038 14	.026 9	.018 7	.015 5
.76	.045 15	.052 16	.035 10	.025 9	.020 7
.78	.060 19	.068 18	.045 13	.034 10	.027 9
0.80	1.079 10	1.086 10	1.058 7	1.044 6	1.036 5
.81	.089 10	.096 11	.065 9	.050 7	.041 5
.82	.099 11	.107 12	.074 9	.057 7	.046 6
.83	.110 12	.119 14	.083 11	.064 8	.052 6
.84	.122 13	.133 14	.094 11	.072 9	.058 7
0.85	1.135 14	1.147 15	1.105 13	1.081 10	1.065 8
.86	.149 15	.162 17	.118 14	.091 10	.073 8
.87	.164 16	.179 19	.132 16	.101 13	.081 10
.88	.180 16	.198 19	.148 18	.114 15	.091 10
.89	.196 18	.217 22	.166 19	.129 15	.101 11
0.90	1.214 19	1.239 25	1.185 21	1.144 17	1.112 13
.91	.233 20	.264 28	.206 24	.161 19	.125 15
.92	.253 22	.292 30	.230 27	.180 22	.140 17
.93	.275 24	.322 33	.257 30	.202 26	.157 19
.94	.299 26	.355 36	.287 33	.228 29	.176 22
0.95	1.325 14	1.391 21	1.320 17	1.257 15	1.198 12
.955	.339 15	.412 21	.337 18	.272 16	.210 13
.96	.354 16	.433 22	.355 19	.288 17	.223 13
.965	.370 17	.455 24	.374 20	.305 18	.236 14
.97	.387 17	.479 25	.394 21	.323 19	.250 16
0.975	1.404 18	1.504 26	1.415 23	1.342 20	1.266 17
.98	.422 21	.530 28	.438 25	.362 22	.283 17
.985	.443 25	.558 29	.463 28	.384 24	.300 19
.99	.468 15	.587 15	.491 18	.408 15	.319 12
.9925	.483 16	.602 16	.509 20	.423 15	.331 12
0.995	1.499 7	1.618 5	1.529 7	1.438 7	1.343 6
.996	.506 9	.623 6	.536 8	.445 8	.349 6
.997	.515 10	.629 7	.544 9	.453 9	.355 7
.998	.525 11	.636 6	.553 10	.462 10	.362 8
.999	.536 9	.642 3	.563 9	.472 4	.370 5
0.9995	1.545 14	1.645 2	1.572 13	1.476 6	1.375 7
1.000	.559	.647	.585	.482	.382

TABLE 1a—continued

Values of $\kappa : r_c = 0.25$

$\sin \phi \backslash N$	2	3	4	5	6
0.58	0.954 2	0.975 3	0.992 0	0.996 0	0.999 -1
0.60	0.956 3	0.978 4	0.992 0	0.996 1	0.998 0
.62	.959 3	.982 4	.992 1	.997 1	.998 +1
.64	.962 3	.986 4	.993 2	.998 1	.999 1
.66	.965 4	.990 6	.995 3	.999 2	1.000 1
.68	.969 4	.996 6	.998 4	1.001 3	.001 2
0.70	0.973 5	1.002 7	1.002 6	1.004 4	1.003 3
.72	.978 5	.009 8	.008 7	.008 5	.006 4
.74	.983 5	.017 10	.015 9	.013 7	.010 6
.76	.988 6	.027 11	.024 11	.020 9	.016 7
.78	.994 7	.038 14	.035 12	.029 11	.023 9
0.80	1.001 4	1.052 7	1.047 8	1.040 5	1.032 5
.81	.005 4	.059 8	.055 8	.045 7	.037 5
.82	.009 5	.067 9	.063 8	.052 7	.042 6
.83	.014 5	.076 10	.071 9	.059 8	.048 7
.84	.019 6	.086 10	.080 9	.067 8	.055 7
0.85	1.025 6	1.096 12	1.089 11	1.075 9	1.062 8
.86	.031 7	.108 12	.100 12	.084 11	.070 9
.87	.038 7	.120 14	.112 13	.095 12	.079 9
.88	.045 8	.134 15	.125 13	.107 12	.088 10
.89	.053 9	.149 15	.138 14	.119 14	.098 12
0.90	1.062 10	1.164 16	1.152 16	1.133 14	1.110 13
.91	.072 11	.180 17	.168 18	.147 16	.123 15
.92	.083 12	.197 18	.186 19	.163 18	.138 15
.93	.095 14	.215 20	.205 20	.181 19	.153 17
.94	.109 15	.235 20	.225 20	.200 20	.170 18
0.95	1.124 7	1.255 11	1.245 12	1.220 11	1.188 9
.955	.131 8	.266 11	.257 12	.231 12	.197 10
.96	.139 9	.277 11	.269 13	.243 12	.207 10
.965	.148 9	.288 12	.282 13	.255 13	.217 11
.97	.157 9	.300 13	.295 14	.268 13	.228 11
0.975	1.166 10	1.313 13	1.309 15	1.281 14	1.239 12
.98	.176 11	.326 14	.324 16	.295 15	.251 13
.985	.187 12	.340 15	.340 17	.310 16	.264 15
.99	.199 13	.355 15	.357 20	.326 18	.279 16
.995	.212 9	.370 7	.377 12	.344 10	.295 8
0.9975	1.221 12	1.377 7	1.389 15	1.354 13	1.303 10
1.000	.233	.384	.404	.367	.313

TABLE 1a—*continued*Values of κ : $r_c = 0.3$

$\sin \phi \backslash N$	2	3	4	5	6
0.00	1.000	1.000	1.000	1.000	1.000
.05	.000 0	.000 0	.000 0	.000 0	.000 0
.10	0.998 - 2	0.997 - 3	0.999 - 1	0.998 - 2	.000 0
.15	.994 - 4	.994 - 3	.998 - 1	.997 - 1	.999 - 1
.175	.991 - 3	.993 - 1	.998 0	.996 - 1	.998 - 1
	- 2	- 2	- 1	- 1	0
0.20	0.989	0.991	0.997	0.995	0.998
.22	.986 - 3	.990 - 1	.996 - 1	.995 0	.998 0
.24	.983 - 3	.989 - 1	.996 0	.994 - 1	.997 - 1
.26	.980 - 3	.988 - 1	.995 - 1	.993 - 1	.997 0
.28	.978 - 2	.986 - 2	.994 - 1	.992 - 1	.996 - 1
	- 3	- 2	- 1	0	0
0.30	0.975	0.984	0.993	0.992	0.996
.32	.972 - 3	.983 - 1	.992 - 1	.991 - 1	.996 0
.34	.969 - 3	.982 - 1	.992 0	.990 - 1	.995 - 1
.36	.966 - 3	.981 - 1	.991 - 1	.990 0	.995 0
.38	.963 - 3	.980 - 1	.990 - 1	.990 0	.994 - 1
	- 3	- 1	0	0	0
0.40	0.960	0.979	0.990	0.990	0.994
.42	.957 - 3	.979 0	.989 - 1	.989 - 1	.994 0
.44	.953 - 4	.978 - 1	.988 - 1	.989 0	.993 - 1
.46	.950 - 3	.978 0	.988 0	.989 0	.993 0
.48	.947 - 3	.978 0	.987 - 1	.989 0	.992 - 1
	- 3	0	- 1	0	0
0.50	0.944	0.978	0.986	0.989	0.992
.52	.941 - 3	.978 0	.986 0	.989 0	.992 0
.54	.938 - 3	.978 0	.986 0	.990 + 1	.993 + 1
.56	.935 - 3	.978 0	.986 0	.990 0	.993 0
.58	.933 - 2	.979 + 1	.987 + 1	.991 + 1	.994 + 1
	- 3	1	1	1	1
0.60	0.930	0.980	0.988	0.992	0.995
.62	.928 - 2	.981 1	.989 1	.994 2	.996 1
.64	.926 - 2	.983 2	.991 2	.996 2	.997 1
.66	.924 - 2	.985 2	.994 3	.998 2	.999 2
.68	.922 - 2	.988 3	.997 3	1.001 3	1.001 2
	- 2	3	3	3	3
0.70	0.920	0.991	1.000	1.004	1.004
.72	.919 - 1	.995 4	.005 5	.008 4	.008 4
.74	.918 - 1	1.000 5	.011 6	.012 4	.012 4
.76	.917 - 1	.005 5	.017 6	.017 5	.017 5
.78	.917 0	.012 7	.024 7	.024 7	.023 6
	0	7	8	8	8
0.80	0.917	1.019	1.032	1.032	1.031
.81	.917 0	.023 4	.037 5	.037 5	.036 5
.82	.918 + 1	.028 5	.042 5	.042 5	.041 5
.83	.919 1	.033 5	.047 5	.047 5	.046 5
.84	.920 2	.038 5	.053 6	.053 6	.052 6
	2	6	6	7	6
0.85	0.922	1.044	1.059	1.060	1.058
.86	.924 2	.050 6	.066 7	.067 7	.065 7
.87	.926 2	.057 7	.074 8	.075 8	.072 7
.88	.929 3	.064 7	.083 9	.085 10	.080 8
.89	.934 5	.072 8	.093 10	.095 10	.089 9
	5	9	11	10	10
0.90	0.939	1.081	1.104	1.105	1.099

TABLE 1a—continued

Values of $\kappa : r_c = 0.3$ —continued

$\sin \phi \backslash N$	2	3	4	5	6
0.90	0.939	1.081	1.104	1.105	1.099
.905	.941	.086	.110	.111	.104
.91	.944	.091	.116	.117	.109
.915	.947	.096	.122	.123	.115
.92	.950	.101	.129	.129	.121
0.925	0.953	1.107	1.135	1.136	1.127
.93	.956	.112	.142	.143	.134
.935	.959	.117	.149	.150	.140
.94	.963	.122	.157	.158	.147
.945	.966	.128	.164	.165	.153
0.95	0.970	1.134	1.171	1.173	1.160
.955	.973	.139	.179	.182	.167
.96	.977	.145	.187	.190	.174
.965	.981	.152	.195	.199	.181
.97	.985	.158	.203	.208	.189
0.975	0.989	1.164	1.211	1.217	1.197
.98	.993	.170	.220	.226	.206
.985	.997	.175	.229	.235	.215
.99	1.001	.180	.238	.244	.224
.995	.006	.185	.247	.254	.234
1.000	1.012	1.189	1.256	1.264	1.245

TABLE 1a—continued

Values of $\kappa : r_c = 0.45$

$\sin \phi \backslash N$	2	3	4	5	6
0.00	1.000	1.000	1.000	1.000	1.000
.02	1.000	0.999	.000	.000	.000
.05	1.000	.998	.000	.001	.001
.07	1.000	.998	.000	.001	.002
0.10	0.999	0.997	1.000	1.002	1.003
.12	.998	.996	.000	.002	.004
.14	.998	.995	0.999	.002	.004
.16	.996	.994	0.999	.003	.004
.18	.994	.992	0.998	.003	.005
0.20	0.990	0.990	0.998	1.003	1.005
.22	.986	.988	.997	.003	.006
.24	.980	.986	.997	.003	.006
.26	.974	.983	.996	.003	.006
.28	.967	.980	.995	.003	.007
0.30	0.959	0.977	0.993	1.003	1.007

TABLE 1a—continued

Values of κ : $r_c = 0.45$ —continued

N Sin ϕ	2	3	4	5	6
0.30	0.959	0.977	0.993	1.003	1.007
.32	.950 - 9	.973 - 4	.992 - 1	.003 0	.007 0
.34	.940 -10	.969 - 4	.991 - 1	.002 - 1	.007 0
.36	.929 -11	.965 - 4	.989 - 2	.002 0	.007 0
.38	.918 -11	.960 - 5	.987 - 2	.001 - 1	.007 0
		- 6	- 2	- 1	0
0.40	0.907	0.954	0.985	1.000	1.007
.42	.896 -11	.948 - 6	.983 - 2	.998 - 2	.007 0
.44	.885 -11	.942 - 6	.980 - 3	.997 - 1	.006 - 1
.46	.873 -12	.936 - 6	.978 - 2	.996 - 1	.005 - 1
.48	.860 -13	.929 - 7	.975 - 3	.995 - 1	.005 0
		- 6	- 3	- 1	- 1
0.50	0.847	0.923	0.972	0.994	1.004
.52	.834 -13	.916 - 7	.969 - 3	.993 - 1	.004 0
.54	.821 -13	.909 - 7	.965 - 4	.991 - 2	.004 0
.56	.808 -13	.902 - 7	.962 - 3	.990 - 1	.004 + 1
.58	.796 -12	.895 - 7	.958 - 4	.989 - 1	.005 0
		- 7	- 3	- 1	0
0.60	0.783	0.888	0.955	0.988	1.005
.62	.771 -12	.881 - 7	.951 - 4	.987 - 1	.005 0
.64	.758 -13	.874 - 7	.948 - 3	.986 - 1	.005 1
.66	.746 -12	.867 - 7	.944 - 4	.985 - 1	.006 1
.68	.734 -12	.860 - 7	.940 - 4	.984 - 1	.007 0
		- 7	- 4	- 1	0
0.70	0.722	0.853	0.936	0.983	1.007
.72	.712 -10	.846 - 7	.933 - 3	.982 - 1	.007 0
.74	.702 -10	.839 - 7	.929 - 4	.981 - 1	.007 0
.76	.693 - 9	.832 - 7	.926 - 3	.980 - 1	.007 1
.78	.684 - 9	.826 - 6	.923 - 3	.979 - 1	.008 0
		- 8	- 3	- 1	0
0.80	0.676	0.820	0.920	0.978	1.008
.82	.670 - 6	.815 - 5	.917 - 3	.977 - 1	.009 1
.84	.664 - 6	.810 - 5	.914 - 3	.977 0	.010 1
.86	.659 - 5	.807 - 3	.913 - 1	.977 + 1	.011 1
.88	.654 - 5	.806 - 1	.912 - 1	.978 0	.012 2
		- 1	+ 1	2	3
0.90	0.649	0.805	0.913	0.980	1.016
.92	.645 - 4	.805 0	.914 1	.981 1	.020 4
.94	.641 - 4	.806 + 1	.915 1	.983 2	.024 4
.96	.638 - 3	.806 0	.917 2	.986 3	.029 5
.98	.634 - 4	.807 1	.920 3	.990 4	.034 5
		- 3	4	6	6
1.00	0.631	0.808	0.924	0.996	1.040

TABLE 1a—continued

Values of $\kappa : r_c = 0.6$

N Sin ϕ	2	3	4	5	6
0.00	1.000	1.000	1.000	1.000	1.000
.01	.000	.000	.000	.000	.000
.02	.000	.999	.000	.000	.001
.03	.000	.999	.000	.000	.001
.04	.000	.999	.000	.001	.001
0.05	1.000	0.999	1.000	1.001	1.002
.06	.999	.998	.000	.001	.002
.07	.999	.998	.000	.001	.003
.08	.999	.998	.000	.002	.003
.09	.998	.997	.000	.002	.003
0.10	0.997	0.997	1.000	1.002	1.004
.11	.996	.996	0.999	.002	.004
.12	.994	.996	.999	.002	.004
.13	.992	.995	.999	.002	.004
.14	.990	.994	.999	.002	.005
0.15	0.987	0.993	0.998	1.002	1.005
.16	.984	.992	.998	.002	.006
.17	.979	.990	.997	.002	.006
.18	.974	.988	.997	.002	.007
.19	.967	.986	.996	.003	.007
0.20	0.960	0.984	0.996	1.003	1.007
.21	.952	.982	.995	.003	.008
.22	.945	.979	.994	.003	.008
.23	.937	.976	.993	.002	.008
.24	.929	.972	.992	.002	.008
0.25	0.921	0.969	0.990	1.002	1.008
.26	.912	.965	.989	.001	.009
.27	.903	.961	.987	.001	.009
.28	.894	.957	.985	.000	.009
.29	.885	.952	.983	.000	.009
0.30	0.876	0.947	0.981	0.999	1.008
.32	.858	.937	.976	.996	.007
.34	.839	.926	.971	.993	.006
.36	.820	.915	.965	.990	.004
.38	.801	.903	.958	.986	.003
0.40	0.782	0.891	0.949	0.982	1.000
.42	.763	.878	.941	.977	0.998
.44	.744	.864	.932	.972	.995
.46	.726	.851	.923	.966	.991
.48	.708	.837	.914	.960	.987
0.50	0.690	0.824	0.905	0.954	0.983
.52	.672	.810	.895	.947	.979
.54	.655	.797	.885	.940	.974
.56	.639	.783	.875	.933	.969
.58	.623	.770	.864	.925	.964
0.60	0.608	0.757	0.854	0.918	0.959

TABLE 1a—continued
 Values of $\kappa : r_c = 0.6$ —continued

N Sin ϕ	2	3	4	5	6
0.60	0.608	0.757	0.854	0.918	0.959
.62	.594 -14	.744 -13	.844 -10	.910 -8	.953 -6
.64	.581 -13	.731 -13	.834 -10	.902 -8	.947 -6
.66	.568 -13	.719 -12	.824 -10	.894 -8	.941 -6
.68	.556 -12	.707 -11	.814 -10	.886 -8	.935 -6
0.70	0.544	0.696	0.803	0.878	0.929
.72	.534 -10	.685 -11	.793 -10	.869 -9	.922 -7
.74	.524 -10	.674 -11	.783 -10	.860 -9	.915 -7
.76	.514 -10	.663 -11	.774 -9	.852 -8	.909 -6
.78	.504 -10	.653 -10	.764 -10	.844 -8	.902 -7
0.80	0.496	0.644	0.755	0.836	0.896
.82	.488 -8	.635 -9	.746 -9	.828 -8	.889 -7
.84	.480 -8	.626 -9	.737 -9	.820 -8	.882 -7
.86	.473 -7	.618 -8	.729 -8	.813 -7	.876 -6
.88	.465 -8	.610 -8	.722 -7	.806 -7	.870 -6
0.90	0.458	0.603	0.714	0.799	0.864
.92	.451 -7	.596 -7	.707 -7	.792 -7	.858 -6
.94	.444 -7	.589 -7	.700 -7	.785 -7	.852 -6
.96	.437 -7	.582 -7	.694 -6	.779 -6	.846 -6
.98	.430 -7	.575 -6	.687 -6	.773 -6	.840 -6
1.00	0.424	0.569	0.681	0.767	0.834

TABLE 1a—continued
 Values of $\kappa : r_c = 0.7$

N Sin ϕ	2	3	4	5	6
0.00	1.000	1.000	1.000	1.000	1.000
.01	.000	.000	.000	.000	.000
.02	.000	0.999 -1	.000	.000	.000
.03	.000	.999 0	.000	.000	.000
.04	0.999 -1	.999 0	.000	.000	.001
0.05	0.999	0.999	1.000	1.001	1.001
.06	.998 -1	.998 -1	.000	.001	.002
.07	.997 -1	.998 0	.000	.001	.002
.08	.995 -2	.998 0	.000	.001	.002
.09	.991 -4	.997 -1	.000	.001	.002
0.10	0.987	0.996	1.000	1.001	1.003
.11	.981 -6	.994 -2	0.999 -1	.001	.003
.12	.975 -6	.993 -1	.999 0	.001	.003
.13	.968 -7	.991 -2	.998 -1	.001	.003
.14	.961 -7	.988 -3	.997 -1	.000 -1	.004
0.15	0.953	0.984	0.996	1.000	1.004

TABLE 1a—continued

Values of κ : $r_c = 0.7$ —continued

N Sin ϕ	2.	3	4	5	6
0.15	0.953	0.984	0.996	1.000	1.004
.16	.944	.981	.994	.000	.004
.17	.934	.977	.993	0.999	.004
.18	.923	.972	.991	.998	.004
.19	.911	.966	.988	.998	.004
0.20	0.899	0.961	0.986	0.997	1.004
.22	.875	.948	.980	.993	.003
.24	.850	.934	.973	.990	.001
.26	.825	.919	.965	.985	.000
.28	.800	.903	.956	.980	0.997
0.30	0.776	0.887	0.945	0.974	0.994
.32	.753	.870	.933	.967	.989
.34	.730	.853	.921	.959	.984
.36	.707	.836	.909	.950	.977
.38	.685	.819	.896	.941	.971
0.40	0.663	0.801	0.883	0.932	0.963
.42	.642	.784	.869	.921	.956
.44	.622	.766	.855	.910	.948
.46	.602	.749	.841	.898	.940
.48	.583	.732	.826	.887	.931
0.50	0.565	0.716	0.812	0.876	0.922
.52	.549	.700	.798	.864	.912
.54	.533	.684	.785	.853	.902
.56	.518	.669	.771	.841	.892
.58	.504	.655	.757	.830	.882
0.60	0.492	0.641	0.744	.0818	0.872
.62	.479	.627	.732	.807	.862
.64	.468	.614	.719	.795	.853
.66	.457	.601	.707	.784	.843
.68	.447	.589	.694	.772	.833
0.70	0.437	0.577	0.682	0.761	0.823
.72	.427	.566	.671	.751	.813
.74	.418	.555	.660	.740	.803
.76	.409	.544	.649	.729	.793
.78	.401	.534	.638	.719	.784
0.80	0.393	0.524	0.628	0.709	0.775
.82	.385	.514	.618	.700	.766
.84	.378	.505	.608	.691	.756
.86	.371	.497	.599	.681	.747
.88	.364	.488	.590	.672	.738
0.90	0.356	0.480	0.582	0.663	0.730
.92	.350	.472	.573	.655	.722
.94	.344	.465	.565	.647	.714
.96	.338	.458	.557	.639	.706
.98	.331	.451	.550	.631	.698
1.00	0.325	0.444	0.543	0.624	0.692

TABLE 1a—continued

Values of $\kappa: r_c = 0.8$

N $\sin \phi$	2	3	4	5	6
0.000	1.000	1.000	1.000	1.000	1.000
.005	.000	.000	.000	.000	.000
.010	.000	.000	.000	.000	.000
.015	.000	.000	.000	.000	.000
.020	0.999 - 1 0	0.999 - 1 0	.000	.000	.000
0.025	0.999 0	0.999 0	1.000	1.000	1.000
.030	.999 - 1	.999 0	.000	.000	.000
.035	.998 - 1	.999 0	.000	.000	.000
.040	.997 - 2	.999 0	.000	.000	.000
.045	.995 - 2	.999 0	.000	.000	.000
0.050	0.993 - 3	0.999 - 1	1.000	1.000	1.000
.055	.990 - 3	.998 - 1	.999 - 1	.000	.000
.060	.987 - 3	.997 - 1	.999 0	.000	.001 1
.065	.982 - 5	.996 - 1	.999 0	.000	.001 0
.070	.977 - 7	.995 - 2	.999 - 1	.000	.001 0
0.075	0.970 - 7	0.993 - 1	0.998 0	1.000	1.001 0
.080	.963 - 7	.992 - 1	.998 0	.000	.001 0
.085	.956 - 7	.990 - 2	.997 - 1	.000	.001 0
.090	.950 - 8	.987 - 3	.996 - 1	0.999 - 1	.001 0
.095	.942 - 7	.984 - 3	.995 - 1	.999 0	.001 - 1
0.100	0.935 - 15	0.981 - 7	0.994 - 3	0.999 - 2	1.000 0
.11	.920 - 15	.974 - 8	.991 - 4	.997 - 3	.000 - 1
.12	.905 - 15	.966 - 8	.987 - 4	.994 - 3	0.999 - 1
.13	.890 - 15	.957 - 9	.983 - 4	.992 - 2	.998 - 1
.14	.875 - 15	.948 - 10	.978 - 5	.989 - 3	.997 - 1
0.15	0.860 - 16	0.938 - 11	0.972 - 6	0.986 - 4	0.996 - 2
.16	.844 - 16	.927 - 11	.966 - 6	.982 - 4	.994 - 3
.17	.828 - 16	.916 - 11	.960 - 6	.978 - 4	.991 - 3
.18	.812 - 16	.905 - 11	.953 - 7	.974 - 4	.989 - 2
.19	.796 - 16	.894 - 11	.945 - 8	.969 - 5	.986 - 3
0.20	0.780 - 31	0.883 - 22	0.937 - 16	0.964 - 11	0.983 - 7
.22	.749 - 31	.861 - 23	.921 - 17	.953 - 12	.976 - 9
.24	.718 - 31	.838 - 23	.904 - 17	.941 - 12	.967 - 9
.26	.690 - 28	.816 - 22	.887 - 17	.929 - 12	.958 - 9
.28	.662 - 28	.793 - 23	.869 - 18	.915 - 14	.948 - 10
0.30	0.635 - 25	0.771 - 22	0.851 - 19	0.901 - 14	0.937 - 13
.32	.610 - 25	.749 - 22	.832 - 19	.887 - 14	.924 - 13
.34	.586 - 24	.727 - 22	.814 - 18	.872 - 15	.911 - 13
.36	.564 - 22	.706 - 21	.796 - 18	.856 - 16	.898 - 13
.38	.542 - 22	.686 - 20	.778 - 18	.841 - 15	.884 - 14
0.40	0.521 - 20	0.667 - 19	0.760 - 18	0.825 - 16	.0871 - 14
.42	.501 - 18	.648 - 19	.742 - 18	.809 - 16	.857 - 14
.44	.483 - 18	.629 - 19	.725 - 17	.793 - 16	.843 - 14
.46	.466 - 17	.611 - 18	.708 - 17	.778 - 15	.829 - 14
.48	.449 - 17	.594 - 17	.691 - 17	.762 - 16	.815 - 14
.50	.434 - 15	.578 - 16	.675 - 16	.747 - 15	.801 - 14

TABLE 1a—continued

Values of κ : $r_c = 0.8$ —continued

N Sin ϕ	2	3	4	5	6
0.50	0.434	0.578	0.675	0.747	0.801
.52	.420 -14	.562 -16	.659 -16	.732 -15	.787 -14
.54	.407 -13	.547 -15	.643 -16	.718 -14	.772 -15
.56	.396 -11	.533 -14	.628 -15	.703 -15	.759 -13
.58	.385 -11	.520 -13	.614 -14	.689 -14	.746 -13
	-10	-13	-13	-14	-13
0.60	0.375	0.507	0.601	0.675	0.733
.62	.366 -9	.494 -13	.588 -13	.662 -13	.720 -13
.64	.357 -9	.482 -12	.576 -12	.650 -12	.708 -12
.66	.348 -9	.470 -12	.564 -12	.638 -12	.696 -12
.68	.340 -8	.460 -10	.552 -12	.626 -12	.685 -11
	-8	-11	-12	-12	-11
0.70	0.332	0.449	0.540	.614	.674
.72	.324 -8	.439 -10	.529 -11	.603 -11	.663 -11
.74	.317 -7	.430 -9	.519 -10	.593 -10	.652 -11
.76	.309 -8	.420 -10	.509 -10	.583 -10	.642 -10
.78	.302 -7	.411 -9	.499 -10	.573 -10	.632 -10
	-6	-8	-9	-9	-9
0.80	0.296	0.403	0.490	0.562	0.623
.82	.289 -7	.395 -8	.481 -9	.553 -9	.613 -10
.84	.283 -6	.387 -8	.472 -9	.544 -9	.603 -10
.86	.277 -6	.379 -8	.463 -9	.536 -8	.595 -8
.88	.271 -6	.371 -8	.455 -8	.528 -8	.586 -9
	-6	-7	-7	-9	-8
0.90	0.265	0.364	0.448	0.519	0.578
.92	.260 -5	.357 -7	.440 -8	.511 -8	.570 -8
.94	.254 -6	.350 -7	.433 -7	.503 -8	.562 -8
.96	.249 -5	.344 -6	.425 -8	.496 -7	.554 -8
.98	.244 -5	.338 -6	.418 -7	.488 -8	.546 -8
	-5	-6	-7	-7	-7
1.00	0.239	0.332	0.411	0.481	0.539

TABLE 1a—continued

Values of $\kappa : r_c = 0.9$

N Sin ϕ	2	3	4	5	6
0.0000	1.000	1.000	1.000	1.000	1.000
.0025	1.000	1.000	.000	.000	.000
.0050	1.000	1.000	.000	.000	.000
.0075	0.999 - 1	1.000	.000	.000	.000
.0100	0.999 0	0.999 - 1	.000	.000	.000
0.0125	0.999 - 1	0.999 0	1.000	1.000	1.000
.0150	.998 - 1	.999 0	1.000	1.000	.000
.0175	.997 - 1	.999 0	0.999 - 1	0.999 - 1	.000
.0200	.996 - 1	.999 0	.999 0	.999 0	.000
.0225	.993 - 3	.998 - 1	.999 0	.999 0	.000
0.0250	0.990 - 4	0.998 - 1	0.999 0	0.999 0	1.000
.0275	.986 - 5	.997 - 1	.999 0	.999 0	1.000
.0300	.981 - 5	.996 - 2	.999 - 1	.999 0	.000
.0325	.976 - 6	.994 - 2	.998 - 1	.999 0	.000
.0350	.970 - 7	.993 - 2	.998 - 1	.999 0	0.999 - 1
0.0375	0.963 - 7	0.991 - 3	0.997 - 1	0.999 0	0.999 0
.0400	.956 - 7	.988 - 3	.996 - 1	.999 - 1	.999 0
.0425	.949 - 7	.985 - 3	.995 - 1	.998 0	.999 - 1
.0450	.942 - 8	.982 - 4	.994 - 2	.998 - 1	.998 0
.0475	.934 - 8	.978 - 3	.992 - 1	.997 0	.998 0
0.0500	0.926 - 17	0.975 - 9	0.991 - 4	0.997 - 2	0.998 0
.055	.909 - 17	.966 - 9	.987 - 5	.995 - 2	.998 - 1
.060	.892 - 16	.957 - 10	.982 - 5	.993 - 3	.997 - 2
.065	.876 - 16	.947 - 10	.977 - 6	.990 - 4	.995 - 2
.070	.860 - 15	.937 - 11	.971 - 7	.986 - 4	.993 - 2
0.075	0.845 - 15	0.926 - 11	0.964 - 7	0.982 - 4	0.991 - 2
.080	.830 - 15	.915 - 10	.957 - 7	.978 - 5	.989 - 4
.085	.815 - 15	.905 - 11	.950 - 7	.973 - 5	.985 - 3
.090	.800 - 14	.894 - 11	.943 - 8	.968 - 6	.982 - 4
.095	.786 - 14	.883 - 11	.935 - 9	.962 - 5	.978 - 3
0.100	0.772 - 13	0.872 - 11	0.926 - 8	0.957 - 6	0.975 - 5
.105	.759 - 13	.861 - 11	.918 - 8	.951 - 5	.970 - 4
.110	.746 - 13	.850 - 10	.910 - 8	.946 - 6	.966 - 5
.115	.733 - 12	.840 - 10	.902 - 8	.940 - 6	.961 - 5
.120	.721 - 11	.830 - 10	.894 - 9	.934 - 6	.956 - 5
0.125	0.710 - 11	0.820 - 10	0.885 - 8	0.928 - 6	0.951 - 5
.130	.699 - 11	.810 - 9	.877 - 8	.922 - 7	.946 - 5
.135	.688 - 10	.801 - 9	.869 - 8	.915 - 7	.941 - 5
.140	.678 - 9	.792 - 9	.861 - 8	.908 - 7	.936 - 5
.145	.669 - 10	.783 - 9	.853 - 8	.901 - 7	.931 - 5
0.150	0.659 - 18	0.774 - 17	0.845 - 15	0.894 - 13	0.926 - 11
.16	.641 - 18	.757 - 17	.830 - 15	.881 - 13	.915 - 11
.17	.623 - 18	.740 - 16	.815 - 15	.868 - 13	.904 - 11
.18	.605 - 16	.724 - 15	.800 - 14	.855 - 13	.893 - 11
.19	.589 - 15	.709 - 15	.786 - 14	.842 - 13	.882 - 11
0.20	0.574	0.693	0.772	0.829	0.871

TABLE 1a—continued
 Values of $\kappa : r_c = 0.9$ —continued

$\sin \phi \backslash N$	2	3	4	5	6
0.20	0.574	0.693	0.772	0.829	0.871
.21	.558 -16	.678 -15	.759 -13	.817 -12	.860 -11
.22	.543 -15	.664 -14	.746 -13	.805 -12	.849 -11
.23	.529 -14	.651 -13	.734 -12	.793 -12	.839 -10
.24	.515 -14	.639 -12	.721 -13	.782 -11	.828 -11
0.25	0.502	0.627	0.709	0.771	0.818
.26	.490 -12	.615 -12	.697 -12	.760 -11	.807 -11
.27	.477 -13	.603 -12	.686 -11	.748 -12	.797 -10
.28	.465 -12	.591 -12	.674 -12	.737 -11	.786 -11
.29	.454 -11	.579 -12	.663 -11	.727 -10	.776 -10
0.30	0.443	0.568	0.652	0.716	0.766
.32	.422 -21	.547 -21	.630 -22	.695 -21	.746 -20
.34	.403 -19	.527 -20	.610 -20	.675 -20	.726 -20
.36	.385 -18	.508 -19	.590 -20	.656 -19	.708 -18
.38	.368 -17	.489 -19	.572 -18	.637 -19	.690 -18
0.40	0.351	0.471	0.554	0.619	0.673
.42	.336 -15	.454 -17	.537 -17	.601 -18	.655 -18
.44	.322 -14	.438 -16	.520 -17	.585 -16	.638 -17
.46	.309 -13	.423 -15	.505 -15	.569 -16	.622 -16
.48	.298 -11	.410 -13	.490 -15	.553 -16	.607 -15
0.50	0.289	0.397	0.476	0.539	0.593
.52	.280 -9	.385 -12	.462 -14	.525 -14	.579 -14
.54	.271 -9	.374 -11	.450 -12	.512 -13	.565 -14
.56	.263 -8	.363 -11	.438 -12	.499 -13	.552 -13
.58	.256 -7	.353 -10	.426 -12	.487 -12	.540 -12
0.60	0.250	0.343	0.416	0.476	0.528
.62	.243 -7	.334 -9	.405 -11	.465 -11	.516 -12
.64	.237 -6	.325 -9	.395 -10	.454 -11	.505 -11
.66	.231 -6	.316 -9	.386 -9	.444 -10	.494 -11
.68	.225 -6	.309 -7	.377 -9	.434 -10	.484 -10
0.70	0.219	0.301	0.368	0.425	0.475
.72	.213 -6	.294 -7	.359 -9	.416 -9	.466 -9
.74	.208 -5	.287 -7	.351 -8	.407 -9	.457 -9
.76	.203 -5	.280 -7	.343 -8	.398 -9	.448 -9
.78	.198 -5	.273 -7	.336 -7	.390 -8	.440 -8
0.80	0.193	0.267	0.329	0.383	0.432
.82	.189 -4	.261 -6	.323 -6	.375 -8	.424 -8
.84	.184 -5	.255 -6	.316 -7	.368 -7	.416 -8
.86	.180 -4	.249 -6	.310 -6	.362 -6	.409 -7
.88	.175 -5	.244 -5	.304 -6	.356 -6	.403 -6
0.90	0.171	0.239	0.298	0.350	0.397
.92	.167 -4	.234 -5	.293 -5	.344 -6	.391 -6
.94	.164 -3	.230 -4	.287 -6	.338 -6	.385 -6
.96	.160 -4	.225 -5	.282 -5	.333 -5	.379 -6
.98	.157 -3	.221 -4	.277 -5	.327 -6	.373 -6
1.00	0.154	0.217	0.272	0.322	0.367

TABLE 1a—continued

Values of $\kappa : r_c = 0.95$

$N \backslash \sin \phi$	2	3	4	5	6
0.000	1.000	1.000	1.000	1.000	1.000
.001	.000	.000	.000	.000	.000
.002	.000	.000	.000	.000	.000
.003	.000	.000	.000	.000	.000
.004	.000	.000	.000	.000	.000
0.005	1.000	1.000	1.000	1.000	1.000
.006	1.000	.000	.000	.000	.000
.007	0.999 - 1	.000	.000	.000	.000
.008	.998 - 1	.000	.000	.000	.000
.009	.997 - 1	.000 - 1	.000 - 1	.000	.000
0.010	0.996 - 2	0.999 0	0.999 0	1.000	1.000
.011	.994 - 2	.999 - 1	.999 0	.000	.000
.012	.991 - 3	.998 - 1	.999 0	.000	.000
.013	.987 - 3	.998 0	.999 0	.000	.000
.014	.984 - 4	.997 - 1	.999 0	.000	.000
0.015	0.980 - 5	0.996 - 2	0.999 - 1	1.000	1.000
.016	.975 - 5	.994 - 2	.998 - 1	1.000 - 1	.000
.017	.969 - 6	.993 - 1	.998 0	0.999 0	.000
.018	.964 - 5	.991 - 2	.997 - 1	.999 - 1	.000
.019	.958 - 6	.989 - 3	.996 - 1	.998 0	.000
0.020	0.952 - 13	0.986 - 5	0.995 - 2	0.998 - 1	1.000 - 1
.022	.939 - 13	.981 - 5	.993 - 2	.997 - 1	0.999 0
.024	.926 - 13	.975 - 6	.991 - 2	.996 - 1	.999 - 1
.026	.913 - 13	.967 - 8	.987 - 4	.995 - 2	.998 - 1
.028	.900 - 14	.960 - 8	.983 - 4	.993 - 2	.997 - 1
0.030	0.886 - 13	0.952 - 8	0.979 - 4	0.991 - 3	0.996 - 2
.032	.873 - 13	.944 - 8	.975 - 4	.988 - 3	.994 - 2
.034	.860 - 13	.935 - 9	.970 - 5	.985 - 3	.992 - 1
.036	.847 - 13	.926 - 9	.964 - 6	.982 - 4	.991 - 3
.038	.834 - 12	.917 - 9	.958 - 6	.978 - 4	.988 - 2
0.040	0.822 - 11	0.908 - 9	0.952 - 7	0.974 - 4	0.986 - 3
.042	.811 - 12	.899 - 9	.945 - 7	.970 - 4	.983 - 2
.044	.799 - 11	.890 - 9	.939 - 6	.966 - 4	.981 - 3
.046	.788 - 11	.881 - 9	.932 - 7	.962 - 4	.978 - 4
.048	.777 - 10	.873 - 8	.926 - 6	.957 - 5	.974 - 3
0.050	0.767 - 10	0.864 - 8	0.920 - 7	0.952 - 5	0.971 - 4
.052	.757 - 10	.856 - 8	.913 - 7	.947 - 5	.967 - 4
.054	.747 - 9	.847 - 9	.906 - 7	.942 - 5	.963 - 4
.056	.738 - 9	.839 - 8	.899 - 7	.937 - 5	.959 - 4
.058	.729 - 9	.831 - 8	.892 - 6	.931 - 5	.955 - 4
0.060	0.720 - 9	0.823 - 8	0.886 - 7	0.926 - 5	0.951 - 4
.062	.711 - 8	.815 - 8	.879 - 6	.921 - 6	.947 - 4
.064	.703 - 8	.807 - 8	.873 - 6	.915 - 5	.943 - 4
.066	.695 - 8	.799 - 7	.866 - 6	.910 - 6	.939 - 4
.068	.687 - 8	.792 - 7	.860 - 6	.904 - 5	.935 - 4
0.070	0.679	0.785	0.854	0.899	0.931

TABLE 1a—continued

Values of $\kappa: r_c = 0.95$ —continued

$\sin \phi \backslash N$	2	3	4	5	6
0.070	0.679 -18	0.785 -17	0.854 -16	0.899 -13	0.931 -11
.075	.661 -17	.768 -17	.838 -16	.886 -13	.920 -13
.080	.644 -16	.751 -15	.823 -15	.873 -13	.907 -11
.085	.628 -15	.736 -15	.808 -14	.860 -13	.896 -11
.090	.613 -15	.721 -14	.794 -13	.847 -12	.885 -11
0.095	0.598 -13	0.707 -14	0.781 -13	0.835 -12	0.874 -10
.100	.585 -13	.693 -13	.768 -13	.823 -12	.864 -11
.105	.572 -12	.680 -12	.755 -12	.811 -11	.853 -10
.110	.560 -11	.668 -12	.743 -11	.800 -11	.843 -10
.115	.549 -11	.656 -11	.732 -11	.789 -11	.833 -10
0.120	0.538 -10	0.645 -11	0.721 -11	0.778 -10	0.823 -10
.125	.528 -10	.634 -10	.710 -10	.768 -10	.813 -10
.130	.518 -10	.624 -10	.700 -10	.758 -10	.803 -9
.135	.508 -9	.614 -10	.690 -10	.748 -9	.794 -9
.140	.499 -9	.604 -9	.680 -9	.739 -9	.785 -9
0.145	0.490 -9	0.595 -9	0.671 -9	0.730 -9	0.776 -9
.150	.481 -16	.586 -16	.662 -18	.721 -18	.767 -17
.16	.465 -15	.570 -16	.644 -16	.703 -16	.750 -16
.17	.450 -14	.554 -15	.628 -15	.687 -15	.734 -15
.18	.436 -13	.539 -14	.613 -15	.672 -15	.719 -15
.19	.423 -13	.525 -13	.598 -14	.657 -14	.704 -14
0.20	0.410 -12	0.512 -13	0.584 -13	0.643 -14	0.690 -13
.21	.398 -11	.499 -12	.571 -12	.629 -13	.677 -13
.22	.387 -12	.487 -11	.559 -12	.616 -12	.664 -12
.23	.375 -10	.476 -11	.547 -12	.604 -12	.652 -12
.24	.365 -10	.465 -11	.535 -11	.592 -11	.640 -11
0.25	0.355 -10	0.454 -10	0.524 -10	0.581 -11	0.629 -11
.26	.345 -9	.444 -10	.514 -10	.570 -10	.618 -11
.27	.336 -9	.434 -10	.504 -10	.560 -10	.607 -11
.28	.327 -9	.424 -9	.494 -10	.550 -10	.596 -10
.29	.318 -8	.415 -9	.484 -9	.540 -10	.586 -9
0.30	0.310 -8	0.406 -8	0.475 -9	0.530 -10	0.577 -10
.31	.302 -8	.398 -9	.466 -9	.520 -9	.567 -9
.32	.294 -14	.389 -16	.457 -17	.511 -17	.558 -18
.34	.280 -14	.373 -15	.440 -16	.494 -17	.540 -17
.36	.266 -13	.358 -14	.424 -15	.477 -15	.523 -17
.38	.253 -11	.344 -13	.409 -14	.462 -16	.506 -15
0.40	0.242 -11	0.331 -13	0.395 -14	0.446 -13	0.491 -15
.42	.231 -9	.318 -11	.381 -13	.433 -14	.476 -14
.44	.222 -9	.307 -11	.368 -12	.419 -12	.462 -13
.46	.213 -8	.296 -10	.356 -11	.407 -13	.449 -12
.48	.205 -7	.286 -10	.345 -11	.394 -11	.437 -12
0.50	0.198	0.276	0.334	0.383	0.425

TABLE 1a—continued
 Values of $\kappa: r_c = 0.95$ —continued

$\text{Sin } \phi \backslash N$	2	3	4	5	6
0.50	0.198 — 6	0.276 — 9	0.334 — 9	0.383 — 11	0.425 — 12
.52	.192 — 6	.267 — 8	.325 — 10	.372 — 10	.413 — 11
.54	.186 — 5	.259 — 7	.315 — 9	.362 — 10	.402 — 10
.56	.181 — 5	.252 — 7	.306 — 8	.352 — 9	.392 — 10
.58	.176 — 5	.245 — 7	.298 — 8	.343 — 8	.382 — 8
0.60	0.171 — 5	0.238 — 7	0.290 — 8	0.335 — 8	0.374 — 9
.62	.166 — 4	.231 — 6	.282 — 7	.327 — 8	.365 — 8
.64	.162 — 4	.225 — 6	.275 — 7	.319 — 8	.357 — 8
.66	.158 — 4	.219 — 6	.268 — 7	.311 — 8	.349 — 8
.68	.154 — 4	.213 — 5	.262 — 7	.304 — 7	.341 — 7
0.70	0.150 — 4	0.208 — 6	0.255 — 6	0.297 — 7	0.334 — 7
.72	.146 — 4	.202 — 5	.249 — 6	.290 — 6	.327 — 7
.74	.142 — 4	.197 — 5	.243 — 5	.284 — 6	.320 — 6
.76	.138 — 3	.192 — 4	.238 — 5	.278 — 6	.314 — 6
.78	.135 — 4	.188 — 5	.233 — 5	.272 — 5	.308 — 6
0.80	0.131 — 3	0.183 — 4	0.228 — 5	0.267 — 5	0.302 — 5
.82	.128 — 3	.179 — 4	.223 — 5	.262 — 5	.297 — 6
.84	.125 — 3	.175 — 4	.218 — 5	.257 — 5	.291 — 5
.86	.122 — 3	.171 — 4	.213 — 5	.252 — 5	.286 — 5
.88	.119 — 3	.167 — 3	.209 — 4	.247 — 5	.281 — 5
0.90	0.116 — 2	0.164 — 4	0.205 — 4	0.242 — 4	0.276 — 5
.92	.114 — 3	.160 — 3	.201 — 4	.238 — 4	.271 — 4
.94	.111 — 2	.157 — 3	.197 — 3	.234 — 5	.267 — 4
.96	.109 — 3	.154 — 3	.194 — 4	.229 — 3	.263 — 5
.98	.106 — 2	.151 — 3	.190 — 3	.226 — 4	.258 — 4
1.00	0.104	0.148	0.187	0.222	0.254

TABLE 1a—continued

Values of $\kappa : r_c = 0.975$

$N \backslash \sin \phi$	2	3	4	5	6
0.0000	1.000	1.000	1.000	1.000	1.000
.0005	.000	.000	.000	.000	.000
.0010	.000	.000	.000	.000	.000
.0015	.000	.000	.000	.000	.000
.0020	.000	.000	.000	.000	.000
0.0025	1.000	1.000	1.000	1.000	1.000
.0030	0.999 - 1	1.000	.000	.000	.000
.0035	.999 0	1.000	.000	.000	.000
.0040	.999 0	1.000	.000	.000	.000
.0045	.997 - 2	0.999 - 1	.000	.000	.000
0.0050	0.995	0.999 0	1.000	1.000	1.000
.0055	.993 - 2	.999 0	1.000	.000	.000
.0060	.990 - 3	.999 0	1.000	.000	.000
.0065	.987 - 3	.998 - 1	0.999 - 1	.000	.000
.0070	.983 - 4	.997 - 1	0.999 0	.000	.000
0.0075	0.978	0.995 - 2	0.999 - 1	1.000	1.000
.0080	.973 - 5	.994 - 1	.998 - 1	.000	.000
.0085	.968 - 5	.992 - 2	.998 0	.000	.000
.0090	.962 - 6	.990 - 2	.997 - 1	.000	.000
.0095	.956 - 6	.987 - 3	.996 - 1	.000 - 1	.000
0.0100	0.950	0.985	0.995	0.999	1.000
.012	.923 - 27	.973 - 12	.990 - 5	.997 - 2	0.999 - 1
.014	.896 - 27	.958 - 15	.983 - 7	.993 - 4	.997 - 2
.016	.869 - 27	.941 - 17	.973 - 10	.988 - 5	.994 - 3
.018	.844 - 25	.924 - 17	.962 - 11	.981 - 7	.990 - 4
0.020	0.819	0.905	0.950	0.973	0.985
.022	.796 - 23	.887 - 18	.937 - 13	.964 - 9	.980 - 5
.024	.774 - 22	.869 - 18	.923 - 14	.955 - 9	.973 - 7
.026	.754 - 20	.852 - 17	.910 - 13	.945 - 10	.966 - 7
.028	.735 - 19	.835 - 17	.896 - 14	.934 - 11	.958 - 8
0.030	0.717	0.819	0.882	0.923	0.949
.032	.700 - 17	.804 - 15	.869 - 13	.912 - 11	.940 - 9
.034	.684 - 16	.789 - 15	.856 - 13	.902 - 10	.931 - 9
.036	.670 - 14	.775 - 14	.844 - 12	.891 - 11	.922 - 9
.038	.656 - 14	.761 - 14	.831 - 13	.880 - 11	.914 - 8
0.040	0.643	0.748	0.819	0.869	0.905
.042	.630 - 13	.736 - 12	.807 - 12	.858 - 11	.895 - 10
.044	.618 - 12	.724 - 12	.795 - 12	.848 - 10	.886 - 9
.046	.607 - 11	.713 - 11	.784 - 11	.838 - 10	.877 - 9
.048	.597 - 10	.701 - 12	.774 - 10	.828 - 10	.869 - 8
0.050	0.586	0.690	0.764	0.819	0.860
.052	.576 - 10	.680 - 10	.754 - 10	.809 - 10	.851 - 9
.054	.567 - 9	.671 - 9	.745 - 9	.800 - 9	.842 - 9
.056	.558 - 9	.661 - 10	.735 - 10	.792 - 8	.834 - 8
0.058	0.549	0.652	0.726	0.783	0.826

TABLE 1a—continued

Values of $\kappa : r_c = 0.975$ —continued

N $\sin \phi$	2	3	4	5	6
0.058	0.549 — 8	0.652 — 9	0.726 — 9	0.783 — 9	0.826 — 8
.060	.541 — 19	.643 — 20	.717 — 20	.774 — 20	.818 — 19
.065	.522 — 17	.623 — 19	.697 — 20	.754 — 19	.799 — 18
.070	.505 — 16	.604 — 17	.677 — 17	.735 — 18	.781 — 17
0.075	0.489 — 14	0.587 — 16	0.660 — 17	0.717 — 17	0.764 — 17
.080	.475 — 14	.571 — 14	.643 — 15	.700 — 16	.747 — 15
.085	.461 — 12	.557 — 14	.628 — 15	.684 — 14	.732 — 15
.090	.449 — 12	.543 — 13	.613 — 13	.670 — 14	.717 — 14
.095	.437 — 11	.530 — 12	.600 — 13	.656 — 13	.703 — 14
0.100	0.426 — 10	0.518 — 11	0.587 — 12	0.643 — 12	0.689 — 12
.105	.416 — 9	.507 — 11	.575 — 11	.631 — 12	.677 — 12
.110	.407 — 9	.496 — 10	.564 — 11	.619 — 12	.665 — 12
.115	.398 — 9	.486 — 10	.553 — 11	.608 — 11	.653 — 12
.120	.389 — 8	.476 — 9	.542 — 10	.596 — 10	.642 — 10
0.125	0.381 — 8	0.467 — 8	0.532 — 9	0.586 — 10	0.632 — 10
.130	.373 — 7	.459 — 8	.523 — 9	.576 — 9	.622 — 10
.135	.366 — 7	.451 — 8	.514 — 9	.567 — 9	.612 — 10
.140	.359 — 6	.443 — 8	.505 — 9	.559 — 8	.603 — 9
.145	.353 — 7	.435 — 7	.497 — 7	.550 — 8	.594 — 9
0.150	0.346 — 6	0.428 — 7	0.490 — 8	0.542 — 8	0.585 — 8
.155	.340 — 6	.421 — 7	.482 — 8	.534 — 8	.577 — 8
.160	.334 — 6	.414 — 7	.475 — 7	.526 — 8	.569 — 8
.165	.328 — 6	.408 — 6	.468 — 7	.518 — 8	.562 — 7
.170	.322 — 5	.401 — 6	.462 — 7	.511 — 7	.554 — 8
0.175	0.317 — 6	0.395 — 5	0.455 — 6	0.504 — 6	0.547 — 7
.180	.311 — 5	.390 — 6	.449 — 6	.498 — 6	.540 — 7
.185	.306 — 5	.384 — 6	.443 — 6	.491 — 7	.534 — 6
.190	.301 — 5	.379 — 5	.437 — 6	.485 — 6	.527 — 7
.195	.296 — 4	.373 — 5	.431 — 5	.479 — 6	.521 — 6
0.200	0.292 — 9	0.368 — 9	0.426 — 11	0.473 — 10	0.515 — 12
.21	.283 — 8	.359 — 9	.415 — 10	.463 — 11	.503 — 11
.22	.275 — 9	.350 — 9	.405 — 9	.452 — 10	.492 — 10
.23	.266 — 8	.341 — 8	.396 — 9	.442 — 9	.482 — 10
.24	.258 — 7	.333 — 9	.387 — 9	.433 — 9	.472 — 10
0.25	0.251 — 7	0.324 — 7	0.378 — 8	0.424 — 9	0.462 — 9
.26	.244 — 7	.317 — 8	.370 — 8	.415 — 9	.453 — 9
.27	.237 — 7	.309 — 8	.362 — 8	.406 — 9	.444 — 9
.28	.230 — 6	.302 — 7	.354 — 8	.398 — 8	.436 — 9
.29	.224 — 6	.295 — 7	.347 — 7	.390 — 7	.427 — 8
0.30	0.2174 — 58	0.288 — 6	0.340 — 7	0.383 — 8	0.419 — 8
.31	.2116 — 57	.282 — 6	.333 — 7	.375 — 7	.411 — 7
.32	.2059 — 54	.276 — 6	.326 — 6	.368 — 7	.404 — 7
.33	.2005 — 53	.270 — 6	.320 — 6	.361 — 7	.397 — 7
.34	.1952 — 50	.264 — 6	.313 — 6	.354 — 6	.390 — 7
0.35	0.1902	0.258	0.307	0.348	0.383

TABLE 1a—continued
 Values of $\kappa : r_c = 0.975$ —continued

$\sin \phi \backslash N$	2	3	4	5	6
0.35	0.1902	0.258	0.307	0.348	0.383
.36	.1855 ⁻⁴⁷	.253 ⁻⁵	.301 ⁻⁶	.341 ⁻⁷	.376 ⁻⁷
.37	.1809 ⁻⁴⁶	.248 ⁻⁵	.295 ⁻⁶	.336 ⁻⁵	.370 ⁻⁶
.38	.1766 ⁻⁴³	.243 ⁻⁵	.290 ⁻⁵	.330 ⁻⁶	.364 ⁻⁶
.39	.1725 ⁻⁴¹	.238 ⁻⁵	.284 ⁻⁶	.325 ⁻⁵	.358 ⁻⁶
0.40	0.1686	0.233	0.279	0.319	0.352
.41	.1648 ⁻³⁸	.228 ⁻⁵	.274 ⁻⁵	.313 ⁻⁶	.346 ⁻⁶
.42	.1612 ⁻³⁶	.224 ⁻⁴	.269 ⁻⁵	.308 ⁻⁵	.340 ⁻⁶
.43	.1577 ⁻³⁵	.220 ⁻⁴	.264 ⁻⁵	.303 ⁻⁵	.335 ⁻⁵
.44	.1544 ⁻³³	.216 ⁻⁴	.260 ⁻⁴	.298 ⁻⁵	.329 ⁻⁶
0.45	0.1512	0.211	0.255	0.293	0.324
.46	.1482 ⁻³⁰	.208 ⁻³	.251 ⁻⁴	.288 ⁻⁵	.319 ⁻⁵
.47	.1454 ⁻²⁸	.204 ⁻⁴	.246 ⁻⁵	.284 ⁻⁴	.315 ⁻⁴
.48	.1427 ⁻²⁷	.200 ⁻⁴	.242 ⁻⁴	.279 ⁻⁵	.310 ⁻⁵
.49	.1402 ⁻²⁵	.197 ⁻³	.238 ⁻⁴	.275 ⁻⁴	.306 ⁻⁴
0.50	0.1379	0.1935	0.235	0.271	0.301
.51	.1357 ⁻²²	.1905 ⁻³⁰	.231 ⁻³	.267 ⁻⁴	.297 ⁻⁴
.52	.1336 ⁻²¹	.1876 ⁻²⁹	.228 ⁻³	.264 ⁻³	.293 ⁻⁴
.53	.1316 ⁻²⁰	.1847 ⁻²⁹	.224 ⁻⁴	.260 ⁻⁴	.289 ⁻⁴
.54	.1297 ⁻¹⁹	.1818 ⁻²⁹	.221 ⁻³	.256 ⁻⁴	.285 ⁻⁴
0.55	0.1279	0.1790	0.218	0.253	0.281
.56	.1261 ⁻¹⁸	.1762 ⁻²⁸	.215 ⁻³	.249 ⁻⁴	.278 ⁻³
.57	.1244 ⁻¹⁷	.1736 ⁻²⁶	.212 ⁻³	.246 ⁻³	.274 ⁻⁴
.58	.1227 ⁻¹⁷	.1710 ⁻²⁶	.209 ⁻³	.2429 ⁻³	.271 ⁻³
.59	.1211 ⁻¹⁶	.1686 ⁻²⁴	.206 ⁻³	.2396 ⁻³³	.267 ⁻⁴
0.60	0.1195	0.1663	0.2036	0.2365	0.264
.62	.1164 ⁻³¹	.1617 ⁻⁴⁶	.1982 ⁻⁵⁴	.2305 ⁻⁶⁰	.2583 ⁻⁶²
.64	.1133 ⁻³¹	.1573 ⁻⁴⁴	.1931 ⁻⁵¹	.2249 ⁻⁵⁶	.2523 ⁻⁶⁰
.66	.1103 ⁻³⁰	.1531 ⁻⁴²	.1882 ⁻⁴⁹	.2195 ⁻⁵⁴	.2464 ⁻⁵⁹
.68	.1074 ⁻²⁹	.1491 ⁻⁴⁰	.1835 ⁻⁴⁷	.2144 ⁻⁵¹	.2409 ⁻⁵⁵
0.70	0.1046	0.1453	0.1790	0.2096	0.2357
.72	.1019 ⁻²⁷	.1416 ⁻³⁷	.1748 ⁻⁴²	.2049 ⁻⁴⁷	.2306 ⁻⁵¹
.74	.0992 ⁻²⁷	.1380 ⁻³⁶	.1707 ⁻⁴¹	.2004 ⁻⁴⁵	.2258 ⁻⁴⁸
.76	.0965 ⁻²⁷	.1345 ⁻³⁵	.1668 ⁻³⁹	.1961 ⁻⁴³	.2212 ⁻⁴⁶
.78	.0939 ⁻²⁶	.1312 ⁻³³	.1631 ⁻³⁷	.1919 ⁻⁴²	.2169 ⁻⁴³
0.80	0.0914	0.1280	0.1596	0.1879	0.2126
.82	.0890 ⁻²⁴	.1250 ⁻³⁰	.1561 ⁻³⁵	.1841 ⁻³⁸	.2085 ⁻⁴¹
.84	.0867 ⁻²³	.1221 ⁻²⁹	.1528 ⁻³³	.1804 ⁻³⁷	.2046 ⁻³⁹
.86	.0845 ⁻²²	.1194 ⁻²⁷	.1495 ⁻³³	.1769 ⁻³⁵	.2008 ⁻³⁸
.88	.0824 ⁻²¹	.1167 ⁻²⁷	.1464 ⁻³¹	.1735 ⁻³⁴	.1972 ⁻³⁶
0.90	0.0804	0.1141	0.1434	0.1702	0.1937
.92	.0786 ⁻¹⁸	.1117 ⁻²⁴	.1405 ⁻²⁹	.1670 ⁻³²	.1903 ⁻³⁴
.94	.0769 ⁻¹⁷	.1094 ⁻²³	.1378 ⁻²⁷	.1638 ⁻³²	.1871 ⁻³²
.96	.0753 ⁻¹⁶	.1072 ⁻²²	.1352 ⁻²⁶	.1608 ⁻³⁰	.1839 ⁻³²
.98	.0738 ⁻¹⁵	.1050 ⁻²²	.1326 ⁻²⁶	.1578 ⁻³⁰	.1807 ⁻³²
1.00	0.0725	0.1029	0.1301	0.1548	0.1777

TABLE 1b
 Values of b° : $N = 2$

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	14.95	18.27	21.38	31.08	42.57	53.32	70.0	104.9	152.1	218.3
.1	14.19	17.48	20.56	30.09	41.40	52.02	68.6	103.4	150.1	215.9
.2	13.58	16.82	19.87	29.33	40.54	51.12	67.6	102.4	148.8	214.1
.3	13.09	16.27	19.29	28.73	39.86	50.36	66.9	101.8	147.9	212.8
.4	12.69	15.80	18.79	28.23	39.30	49.70	66.2	101.1	147.3	211.7
1.5	12.35	15.40	18.36	27.80	38.83	49.11	65.5	100.2	146.7	211.0
.6	12.06	15.06	17.99	27.40	38.41	48.62	64.9	99.5	145.8	210.0
.7	11.81	14.76	17.66	27.04	38.06	48.21	64.4	98.6	144.7	208.3
.8	11.60	14.50	17.36	26.71	37.73	47.81	63.9	97.8	143.7	206.6
.9	11.41	14.27	17.09	26.41	37.40	47.43	63.4	97.1	142.7	205.0
2.0	11.24	14.06	16.84	26.13	37.12	47.16	63.0	96.4	141.6	203.6
.1	11.09	13.87	16.62	25.87	36.87	46.93	62.6	95.7	140.5	202.2
.2	10.96	13.70	16.42	25.64	36.66	46.73	62.3	95.0	139.6	200.8
.3	10.85	13.55	16.24	25.44	36.47	46.54	62.0	94.4	138.9	199.6
.4	10.75	13.42	16.07	25.25	36.30	46.36	61.8	94.0	138.3	198.7
2.5	10.66	13.30	15.92	25.07	36.13	46.19	61.6	93.8	137.8	197.9
.6	10.58	13.19	15.78	24.91	35.94	46.02	61.4	93.6	137.4	197.2
.7	10.51	13.09	15.65	24.76	35.76	45.85	61.2	93.4	137.0	196.7
.8	10.44	13.00	15.53	24.63	35.61	45.69	61.0	93.2	136.7	196.2
.9	10.38	12.92	15.42	24.51	35.48	45.54	60.9	93.1	136.4	195.9
3.0	10.32	12.85	15.33	24.40	35.36	45.42	60.8	92.9	136.3	195.6
.1	10.27	12.79	15.25	24.30	35.26	45.32	60.7	92.8	136.2	195.4
.2	10.22	12.73	15.17	24.20	35.17	45.25	60.6	92.7	136.1	195.3
.3	10.18	12.68	15.10	24.11	35.10	45.18	60.5	92.6	136.0	195.2
.4	10.14	12.63	15.04	24.03	35.03	45.11	60.5	92.6	136.0	195.2
3.5	10.10	12.58	14.99	23.96	34.95	45.05	60.4	92.6	136.0	195.2
.6	10.06	12.54	14.94	23.89	34.89	44.99	60.4	92.6	136.0	195.3
.7	10.03	12.49	14.89	23.83	34.83	44.93	60.3	92.6	136.0	195.4
.8	10.00	12.45	14.84	23.78	34.78	44.87	60.3	92.6	136.1	195.4
.9	9.97	12.42	14.80	23.73	34.73	44.81	60.2	92.6	136.1	195.5
4.0	9.94	12.38	14.76	23.68	34.69	44.77	60.2	92.6	136.1	195.6
.1	9.91	12.35	14.73	23.64	34.66	44.73	60.2	92.6	136.2	195.7
.2	9.89	12.33	14.70	23.60	34.63	44.70	60.2	92.6	136.2	195.8
.3	9.87	12.30	14.67	23.56	34.60	44.67	60.1	92.6	136.3	195.9
.4	9.85	12.28	14.64	23.52	34.57	44.65	60.1	92.6	136.4	196.1
4.5	9.83	12.25	14.61	23.48	34.54	44.63	60.1	92.6	136.5	196.2

TABLE 1b—continued
Values of b^0 : $N = 2$ —continued

$J \backslash r_0$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
4.5	9.83	12.25	14.61	23.48	34.54	44.63	60.1	92.6	136.5	196.2
.6	9.82 — 1	12.23 — 2	14.58 — 3	23.46 — 2	34.51 — 3	44.60 — 3	60.0 — 1	92.6 0	136.5 0	196.3 1
.7	9.80 — 2	12.21 — 2	14.56 — 2	23.43 — 3	34.48 — 3	44.58 — 2	60.0 0	92.6 0	136.6 1	196.4 1
.8	9.78 — 2	12.19 — 2	14.54 — 2	23.40 — 3	34.45 — 3	44.55 — 3	60.0 0	92.6 0	136.7 1	196.4 0
.9	9.76 — 2	12.17 — 2	14.52 — 2	23.37 — 3	34.42 — 3	44.53 — 2	60.0 0	92.7 1	136.7 0	196.5 1
5.0	9.74	12.15	14.50	23.34	34.39	44.51	59.9	92.7	136.8	196.6
.1	9.73 — 1	12.14 — 1	14.49 — 1	23.31 — 3	34.37 — 2	44.48 — 3	59.9 0	92.7 0	136.8 0	196.7 1
.2	9.72 — 1	12.12 — 2	14.48 — 1	23.29 — 2	34.35 — 2	44.46 — 2	59.9 0	92.7 0	136.9 1	196.7 0
.3	9.71 — 1	12.11 — 1	14.46 — 2	23.27 — 2	34.32 — 3	44.44 — 2	59.9 0	92.7 0	136.9 0	196.8 1
.4	9.70 — 1	12.09 — 2	14.45 — 1	23.25 — 2	34.29 — 3	44.42 — 2	59.9 0	92.7 0	136.9 0	196.8 0
5.5	9.69	12.08	14.44	23.23	34.26	44.40	59.9	92.7	137.0	196.9
.6	9.68 — 1	12.07 — 1	14.43 — 1	23.21 — 2	34.24 — 2	44.38 — 2	59.9 0	92.7 0	137.0 0	196.9 0
.7	9.67 — 1	12.06 — 1	14.42 — 1	23.19 — 2	34.23 — 1	44.36 — 2	59.9 0	92.7 0	137.0 0	196.9 0
.8	9.66 — 1	12.04 — 2	14.41 — 1	23.18 — 1	34.22 — 1	44.34 — 2	59.9 0	92.7 0	137.0 0	197.0 1
.9	9.65 — 1	12.03 — 1	14.40 — 1	23.17 — 1	34.21 — 1	44.32 — 2	59.9 0	92.7 0	137.0 0	197.0 0
6.0	9.64	12.02	14.39	23.16	34.20	44.30	59.9	92.7	137.0	197.0
6.5	9.59	11.97	14.35	23.09	34.13	44.21	59.9	92.7	137.0	197.2
7.0	9.55	11.93	14.31	23.03	34.10	44.16	59.9	92.7	137.0	197.2

TABLE 1b—continued
 Values of b° : $N = 3$

$J \backslash r_0$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	14.82 -80	17.49 -91	19.74 -97	27.69 -127	36.15 -155	43.83 -170	55.59 -195	80.5 -23	115.2 -29	163.5 -40
.1	14.02 -65	16.58 -76	18.77 -81	26.42 -103	34.60 -122	42.13 -136	53.64 -153	78.2 -18	112.3 -25	159.5 -28
.2	13.37 -54	15.82 -65	17.96 -68	25.39 -84	33.38 -97	40.77 -107	52.11 -118	76.4 -13	109.8 -17	156.7 -24
.3	12.83 -46	15.17 -55	17.28 -57	24.55 -67	32.41 -78	39.70 -88	50.93 -98	75.1 -10	108.1 -13	154.3 -20
.4	12.37 -39	14.62 -47	16.71 -48	23.88 -56	31.63 -65	38.82 -71	49.95 -82	74.1 -9	106.8 -12	152.3 -16
1.5	11.98 -33	14.15 -40	16.23 -42	23.32 -48	30.98 -53	38.11 -59	49.13 -68	73.2 -8	105.6 -9	150.7 -12
.6	11.65 -29	13.75 -34	15.81 -37	22.84 -41	30.45 -46	37.52 -52	48.45 -56	72.4 -8	104.7 -8	149.5 -11
.7	11.36 -25	13.41 -29	15.44 -32	22.43 -35	29.99 -41	37.00 -46	47.89 -46	71.6 -7	103.9 -8	148.4 -10
.8	11.11 -23	13.12 -25	15.12 -28	22.08 -31	29.58 -36	36.54 -38	47.43 -38	70.9 -6	103.1 -7	147.4 -9
.9	10.88 -20	12.87 -22	14.84 -25	21.77 -30	29.22 -31	36.16 -32	47.05 -32	70.3 -5	102.4 -7	146.5 -9
2.0	10.68 -18	12.65 -19	14.59 -22	21.47 -26	28.91 -26	35.84 -26	46.73 -28	69.8 -4	101.7 -7	145.6 -8
.1	10.50 -16	12.46 -17	14.37 -20	21.21 -24	28.65 -22	35.58 -24	46.45 -25	69.4 -3	101.0 -6	144.8 -7
.2	10.34 -14	12.29 -15	14.17 -18	20.97 -22	28.43 -22	35.34 -22	46.20 -23	69.1 -3	100.4 -5	144.1 -6
.3	10.20 -12	12.14 -13	13.99 -16	20.75 -20	28.21 -21	35.12 -22	45.97 -21	68.8 -2	99.9 -4	143.5 -5
.4	10.08 -10	12.01 -12	13.83 -14	20.55 -18	28.00 -19	34.92 -18	45.76 -20	68.6 -2	99.5 -3	143.0 -5
2.5	9.98 -9	11.89 -11	13.69 -12	20.37 -16	27.81 -19	34.74 -14	45.56 -18	68.4 -2	99.2 -2	142.5 -5
.6	9.89 -8	11.78 -10	13.57 -11	20.21 -16	27.64 -15	34.60 -14	45.38 -16	68.2 -2	99.0 -2	142.0 -4
.7	9.81 -7	11.68 -9	13.46 -10	20.05 -14	27.49 -12	34.46 -14	45.22 -14	68.0 -2	98.8 -2	141.6 -3
.8	9.74 -7	11.59 -7	13.36 -9	19.91 -13	27.37 -11	34.32 -12	45.08 -12	67.8 -1	98.6 -2	141.3 -3
.9	9.67 -6	11.52 -7	13.27 -9	19.78 -12	27.26 -11	34.20 -12	44.96 -10	67.7 -1	98.4 -1	141.0 -2
3.0	9.61 -6	11.45 -7	13.18 -8	19.66 -12	27.15 -11	34.08 -9	44.86 -10	67.6 -1	98.3 -1	140.8 -2
.1	9.55 -5	11.38 -6	13.10 -7	19.54 -11	27.04 -10	33.99 -9	44.76 -10	67.5 -1	98.2 -1	140.6 -2
.2	9.50 -5	11.32 -6	13.03 -6	19.43 -10	26.94 -10	33.90 -8	44.66 -10	67.4 -1	98.1 0	140.4 -1
.3	9.45 -4	11.26 -5	12.97 -5	19.33 -8	26.84 -9	33.82 -8	44.56 -10	67.3 -1	98.1 -1	140.3 -1
.4	9.41 -4	11.21 -4	12.92 -5	19.25 -8	26.75 -9	33.74 -7	44.46 -9	67.2 0	98.0 0	140.2 -1
3.5	9.37 -4	11.17 -5	12.87 -5	19.17 -7	26.66 -9	33.67 -7	44.37 -9	67.2 -1	98.0 -1	140.1 0
.6	9.33 -3	11.12 -4	12.82 -4	19.10 -7	26.57 -6	33.60 -7	44.28 -8	67.1 -1	97.9 0	140.1 -1
.7	9.30 -3	11.08 -4	12.78 -4	19.03 -7	26.51 -6	33.53 -6	44.20 -8	67.0 0	97.9 -1	140.0 0
.8	9.27 -3	11.04 -4	12.74 -4	18.96 -6	26.45 -6	33.47 -6	44.12 -7	67.0 0	97.8 -1	140.0 -1
.9	9.24 -3	11.00 -3	12.70 -3	18.90 -5	26.39 -6	33.41 -5	44.05 -5	67.0 0	97.7 -1	139.9 0
4.0	9.21 -2	10.97 -3	12.67 -3	18.85 -5	26.33 -6	33.36 -4	44.00 -4	67.0 -1	97.6 -1	139.9 -1
.1	9.19 -2	10.94 -3	12.64 -3	18.80 -5	26.27 -4	33.32 -4	43.96 -3	66.9 0	97.5 0	139.8 0
.2	9.17 -2	10.91 -2	12.61 -3	18.75 -5	26.23 -4	33.28 -4	43.93 -3	66.9 0	97.5 -1	139.8 -1
.3	9.15 -3	10.89 -2	12.58 -3	18.70 -4	26.19 -4	33.24 -4	43.90 -3	66.9 -1	97.4 -0	139.7 0
.4	9.12 -2	10.87 -2	12.55 -2	18.66 -4	26.15 -4	33.20 -4	43.87 -3	66.8 0	97.4 -1	139.7 0
4.5	9.10	10.85	12.53	18.62	26.11	33.16	43.84	66.8	97.3	139.7

TABLE 1b---continued
 Values of b° : $N = 3$ ---continued

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
4.5	9.10	10.85	12.53	18.62	26.11	33.16	43.84	66.8	97.3	139.7
.6	9.08 -- 2	10.83 -- 2	12.50 -- 3	18.58 -- 4	26.07 -- 4	33.12 -- 4	43.81 -- 3	66.8 0	97.3 0	139.7 -- 1
.7	9.07 -- 1	10.80 -- 3	12.48 -- 2	18.54 -- 4	26.03 -- 4	33.08 -- 4	43.78 -- 3	66.7 -- 1	97.3 0	139.6 -- 0
.8	9.06 -- 1	10.78 -- 2	12.46 -- 2	18.51 -- 3	25.99 -- 4	33.04 -- 4	43.76 -- 2	66.7 0	97.3 0	139.6 -- 0
.9	9.04 -- 2	10.76 -- 2	12.44 -- 2	18.48 -- 3	25.95 -- 4	33.01 -- 3	43.74 -- 2	66.7 0	97.2 -- 1	139.5 -- 1
5.0	9.02	10.74	12.43	18.45	25.92	32.99	43.72	66.6	97.2	139.5
.1	9.01 -- 1	10.73 -- 1	12.42 -- 1	18.43 -- 2	25.89 -- 3	32.97 -- 2	43.70 -- 2	66.6 0	97.2 0	139.5 -- 0
.2	9.00 -- 1	10.72 -- 1	12.40 -- 2	18.40 -- 3	25.87 -- 2	32.95 -- 2	43.68 -- 2	66.6 0	97.2 0	139.5 -- 0
.3	8.99 -- 1	10.71 -- 1	12.39 -- 1	18.38 -- 2	25.84 -- 3	32.93 -- 2	43.66 -- 2	66.6 0	97.1 -- 1	139.5 -- 0
.4	8.98 -- 1	10.70 -- 1	12.37 -- 2	18.36 -- 2	25.82 -- 2	32.91 -- 2	43.64 -- 2	66.5 -- 1	97.1 0	139.5 -- 0
5.5	8.97	10.69	12.36	18.34	25.80	32.89	43.63	66.5	97.1	139.4
.6	8.96 -- 1	10.68 -- 1	12.35 -- 1	18.32 -- 2	25.78 -- 2	32.87 -- 2	43.61 -- 2	66.5 0	97.1 0	139.4 -- 0
.7	8.95 -- 1	10.67 -- 1	12.34 -- 1	18.30 -- 2	25.76 -- 2	32.85 -- 2	43.60 -- 1	66.5 0	97.1 -- 1	139.4 -- 0
.8	8.95 0	10.66 -- 1	12.32 -- 2	18.30 -- 3	25.74 -- 2	32.83 -- 2	43.59 -- 1	66.5 0	97.0 0	139.4 -- 0
.9	8.94 -- 1	10.65 -- 1	12.31 -- 1	18.27 -- 2	25.72 -- 2	32.81 -- 2	43.57 -- 2	66.5 -- 1	97.0 0	139.4 -- 0
6.0	8.93	10.64	12.30	18.23	25.71	32.79	43.56	66.4	97.0	139.4
6.5	8.89	10.59	12.26	18.16	25.64	32.70	43.52	66.4	97.0	139.4
7.0	8.86	10.55	12.23	18.10	25.58	32.64	43.49	66.4	97.0	139.3

TABLE 1b—continued

Values of b° : $N = 4$

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	15.39	17.54	19.53	25.87	33.24	39.64	49.20	69.7	98.3	138.6
.1	14.62	16.65	18.53	24.49	31.47	37.67	47.07	67.2	95.1	134.1
.2	13.97	15.90	17.71	23.40	30.09	36.14	45.35	65.2	92.7	131.4
.3	13.42	15.27	17.03	22.50	28.97	34.90	44.01	63.6	90.7	128.9
.4	12.95	14.74	16.47	21.75	28.05	33.89	42.91	62.4	89.1	126.9
1.5	12.55	14.29	16.00	21.12	27.29	33.08	42.03	61.3	87.8	125.3
.6	12.21	13.90	15.58	20.60	26.65	32.38	41.29	60.5	86.8	123.9
.7	11.92	13.56	15.20	20.14	26.12	31.78	40.68	59.7	85.8	122.6
.8	11.67	13.26	14.86	19.76	25.66	31.26	40.13	59.0	84.9	121.4
.9	11.45	13.00	14.55	19.42	25.27	30.82	39.62	58.5	84.1	120.3
2.0	11.26	12.77	14.27	19.13	24.93	30.46	39.19	58.0	83.5	119.2
.1	11.09	12.57	14.02	18.87	24.63	30.15	38.82	57.6	83.0	118.4
.2	10.94	12.39	13.79	18.63	24.37	29.88	38.49	57.2	82.5	117.7
.3	10.81	12.23	13.59	18.42	24.13	29.63	38.21	56.8	82.1	117.1
.4	10.69	12.09	13.41	18.22	23.92	29.39	37.97	56.5	81.7	116.6
2.5	10.59	11.96	13.25	18.04	23.73	29.17	37.76	56.2	81.3	116.1
.6	10.49	11.84	13.10	17.87	23.56	29.00	37.57	55.9	80.9	115.7
.7	10.41	11.74	12.97	17.72	23.40	28.85	37.39	55.7	80.5	115.3
.8	10.33	11.65	12.85	17.58	23.26	28.71	37.22	55.5	80.2	114.9
.9	10.26	11.57	12.74	17.45	23.14	28.57	37.07	55.4	80.0	114.6
3.0	10.20	11.50	12.64	17.34	23.02	28.44	36.94	55.3	79.9	114.3
.1	10.14	11.43	12.55	17.24	22.90	28.31	36.82	55.2	79.7	114.0
.2	10.08	11.36	12.47	17.14	22.80	28.19	36.71	55.0	79.6	113.7
.3	10.03	11.30	12.40	17.05	22.71	28.10	36.61	54.9	79.5	113.5
.4	9.99	11.25	12.33	16.97	22.62	28.01	36.52	54.8	79.4	113.3
3.5	9.95	11.20	12.27	16.89	22.55	27.93	36.43	54.7	79.3	113.1
.6	9.91	11.16	12.21	16.82	22.48	27.85	36.35	54.6	79.2	112.9
.7	9.87	11.11	12.16	16.75	22.41	27.77	36.28	54.4	79.1	112.7
.8	9.83	11.07	12.12	16.69	22.35	27.71	36.22	54.3	79.0	112.5
.9	9.80	11.03	12.08	16.64	22.28	27.65	36.16	54.2	78.9	112.4
4.0	9.77	10.99	12.04	16.59	22.22	27.59	36.10	54.1	78.8	112.3
.1	9.75	10.96	12.00	16.54	22.17	27.55	36.04	54.0	78.6	112.2
.2	9.73	10.93	11.97	16.49	22.12	27.50	35.98	53.9	78.5	112.1
.3	9.71	10.90	11.94	16.45	22.07	27.45	35.92	53.9	78.4	112.0
.4	9.68	10.88	11.91	16.41	22.03	27.40	35.86	53.8	78.3	111.9
4.5	9.66	10.85	11.89	16.38	21.99	27.35	35.81	53.8	78.2	111.8

TABLE 1b—continued
 Values of b° : $N = 4$ —continued

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
4.5	9.66	10.85	11.89	16.38	21.99	27.35	35.81	53.8	78.2	111.8
.6	9.64	10.83	11.86	16.34	21.95	27.30	35.75	53.8	78.2	111.7
.7	9.62	10.80	11.84	16.31	21.91	27.27	35.70	53.7	78.1	111.6
.8	9.61	10.78	11.82	16.28	21.87	27.24	35.66	53.7	78.1	111.5
.9	9.59	10.76	10.80	16.25	21.84	27.21	35.62	53.6	78.0	111.5
5.0	9.57	10.74	11.78	16.22	21.81	27.18	35.58	53.6	78.0	111.5
.1	9.56	10.72	11.77	16.19	21.78	27.15	35.54	53.5	77.9	111.4
.2	9.54	10.71	11.75	16.16	21.76	27.12	35.51	53.5	77.9	111.4
.3	9.53	10.69	11.74	16.14	21.74	27.10	35.48	53.4	77.9	111.3
.4	9.51	10.68	11.72	16.12	21.72	27.09	35.45	53.4	77.8	111.3
5.5	9.50	10.66	11.71	16.10	21.69	27.08	35.43	53.4	77.8	111.3
.6	9.49	10.65	11.70	16.08	21.66	27.06	35.40	53.4	77.7	111.2
.7	9.48	10.64	11.69	16.06	21.64	27.05	35.38	53.4	77.7	111.2
.8	9.47	10.62	11.68	16.04	21.62	27.03	35.36	53.3	77.6	111.2
.9	9.46	10.61	11.67	16.03	21.60	27.02	35.34	53.3	77.6	111.1
6.0	9.45	10.60	11.66	16.02	21.59	27.01	35.32	53.3	77.5	111.1
6.5	9.40	10.55	11.62	15.94	21.51	26.86	35.23	53.2	77.3	111.0
7.0	9.36	10.51	11.59	15.89	21.44	26.81	35.19	53.1	77.3	110.9

TABLE 1b—continued

Values of b° : $N = 5$

$J \backslash r_0$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	15.7	17.6	19.5	25.1	31.7	37.5	45.6	62.9	88.0	123.0
.1	15.0 ⁻⁷	16.8 ⁻⁸	18.5 ⁻¹⁰	23.6 ⁻¹⁵	29.8 ⁻¹⁹	35.4 ⁻²¹	43.4 ⁻²²	60.5 ⁻²⁴	84.8 ⁻³²	118.9 ⁻⁴¹
.2	14.4 ⁻⁶	16.1 ⁻⁷	17.7 ⁻⁸	22.4 ⁻¹²	28.3 ⁻¹⁵	33.7 ⁻¹⁷	41.5 ⁻¹⁹	58.5 ⁻²⁰	82.2 ⁻²⁶	115.7 ⁻³²
.3	13.9 ⁻⁵	15.5 ⁻⁶	17.1 ⁻⁶	21.5 ⁻⁹	27.1 ⁻¹²	32.3 ⁻¹⁴	40.0 ⁻¹⁵	56.8 ⁻¹⁷	80.1 ⁻²¹	113.1 ⁻²⁶
.4	13.5 ⁻⁴	15.0 ⁻⁵	16.5 ⁻⁶	20.7 ⁻⁸	26.1 ⁻¹⁰	31.2 ⁻¹¹	38.8 ⁻¹²	55.5 ⁻¹³	78.5 ⁻¹⁶	111.0 ⁻²¹
1.5	13.1	14.5	16.0	20.0	25.3	30.3	37.8	54.4	77.1	109.3
.6	12.8	14.1 ⁻⁴	15.6 ⁻⁴	19.5 ⁻⁵	24.6 ⁻⁷	29.5 ⁻⁸	37.0 ⁻⁸	53.4 ⁻¹⁰	75.9 ⁻¹²	107.8 ⁻¹⁵
.7	12.5	13.8	15.2 ⁻⁴	19.0	24.0 ⁻⁶	28.9 ⁻⁶	36.4 ⁻⁶	52.6 ⁻⁸	74.9 ⁻¹⁰	106.5 ⁻¹³
.8	12.2	13.5	14.8 ⁻⁴	18.6	23.5 ⁻⁵	28.3 ⁻⁶	35.8 ⁻⁶	51.9 ⁻⁷	74.1 ⁻⁸	105.4 ⁻¹¹
.9	12.0	13.3	14.5	18.3	23.1 ⁻⁴	27.8 ⁻⁴	35.3 ⁻⁵	51.3 ⁻⁶	73.3 ⁻⁷	104.3 ⁻¹⁰
2.0	11.8	13.0	14.2	18.0	22.7	27.4	34.8	50.7	72.6	103.3
.1	11.7	12.8	14.0	17.7	22.4	27.0	34.4	50.2 ⁻⁵	72.0 ⁻⁶	102.4 ⁻⁹
.2	11.5	12.6	13.8	17.4	22.1	26.7	34.0	49.8	71.5 ⁻⁵	101.6 ⁻⁸
.3	11.4	12.5	13.6	17.2	21.9	26.4	33.7	49.4	71.0 ⁻⁵	100.8 ⁻⁸
.4	11.3	12.3	13.4	17.0	21.6	26.2	33.5	49.1	70.6 ⁻⁴	100.2 ⁻⁶
2.5	11.2	12.2	13.3	16.9	21.4	26.0	33.2	48.8	70.2	99.6
.6	11.1	12.1	13.1	16.7	21.3	25.8	33.0	48.5	69.8	99.1 ⁻⁵
.7	11.0	12.0	13.0	16.6	21.1	25.6	32.8	48.3	69.5	98.7
.8	10.9	11.9	12.9	16.4	20.9	25.4	32.6	48.1	69.2	98.3
.9	10.8	11.8	12.8	16.3	20.8	25.3	32.4	47.9	68.9	98.0
3.0	10.8	11.7	12.7	16.2	20.7	25.4	32.3	47.7	68.7	97.6
.1	10.7	11.7	12.6	16.1	20.6	25.0	32.1	47.6	68.5	97.3
.2	10.7	11.6	12.6	16.0	20.5	24.9	32.0	47.4	68.3	97.0
.3	10.6	11.6	12.5	15.9	20.4	24.8	31.9	47.3	68.1	96.8
.4	10.6	11.5	12.4	15.8	20.3	24.7	31.7	47.2	67.9	96.5
3.5	10.5	11.4	12.4	15.7	20.2	24.6	31.6	47.0	67.7	96.3
.6	10.5	11.4	12.3	15.7	20.1	24.5	31.5	46.9	67.6	96.1
.7	10.5	11.4	12.3	15.6	20.0	24.4	31.4	46.8	67.4	95.9
.8	10.4	11.3	12.2	15.6	20.0	24.4	31.4	46.7	67.3	95.7
.9	10.4	11.3	12.2	15.5	19.9	24.3	31.3	46.6	67.2	95.5
4.0	10.4	11.2	12.1	15.4	19.9	24.3	31.2	46.5	67.0	95.4
.1	10.3	11.2	12.1	15.4	19.8	24.2	31.1	46.5	66.9	95.2
.2	10.3	11.2	12.0	15.4	19.8	24.2	31.1	46.4	66.8	95.1
.3	10.3	11.2	12.0	15.3	19.7	24.1	31.0	46.3	66.7	95.0
.4	10.3	11.1	12.0	15.3	19.7	24.1	31.0	46.2	66.6	94.8
4.5	10.2	11.1	12.0	15.2	19.6	24.0	30.9	46.2	66.5	94.7

TABLE 1b—continued
Values of b° : $N = 5$ —continued

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
4.5	10.2	11.1	12.0	15.2	19.6	24.0	30.9	46.2	66.5	94.7
.6	10.2	11.1	12.0	15.2	19.6	24.0	30.9	46.1	66.4	94.6
.7	10.2	11.0	11.9	15.2	19.6	23.9	30.8	46.0	66.4	94.5
.8	10.2	11.0	11.9	15.1	19.5	23.9	30.8	46.0	66.3	94.5
.9	10.2	11.0	11.9	15.1	19.5	23.9	30.7	45.9	66.2	94.4
5.0	10.2	11.0	11.8	15.1	19.4	23.8	30.7	45.9	66.2	94.3
.1	10.1	11.0	11.8	15.0	19.4	23.8	30.7	45.8	66.1	94.2
.2	10.1	11.0	11.8	15.0	19.4	23.8	30.6	45.8	66.0	94.1
.3	10.1	10.9	11.8	15.0	19.4	23.7	30.6	45.7	66.0	94.1
.4	10.1	10.9	11.8	15.0	19.3	23.7	30.6	45.7	65.9	94.0
5.5	10.1	10.9	11.8	15.0	19.3	23.7	30.6	45.6	65.9	94.0
.6	10.1	10.9	11.8	14.9	19.3	23.6	30.5	45.6	65.8	93.9
.7	10.1	10.9	11.7	14.9	19.3	23.6	30.5	45.6	65.8	93.8
.8	10.0	10.9	11.7	14.9	19.3	23.6	30.5	45.5	65.8	93.8
.9	10.0	10.9	11.7	14.9	19.2	23.6	30.5	45.5	65.7	93.7
6.0	10.0	10.9	11.7	14.9	19.2	23.6	30.4	45.4	65.7	93.7
6.5	9.97	10.8	11.6	14.8	19.1	23.5	30.4	45.3	65.5	93.5
7.0	9.94	10.8	11.6	14.7	19.1	23.4	30.3	45.2	65.4	93.3

TABLE 1b—continued

Values of b° : $N = 6$

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	15.9	17.8	19.5	24.7	30.9	36.1	43.5	58.6	80.5	112.2
.1	15.3 ⁻⁶	16.9 ⁻⁹	18.5 ⁻¹⁰	23.2 ⁻¹⁵	28.9 ⁻²⁰	33.9 ⁻²²	41.0 ⁻²⁵	56.0 ⁻²⁶	77.4 ⁻³¹	108.2 ⁻⁴⁰
.2	14.8 ⁻⁵	16.3 ⁻⁶	17.7 ⁻⁸	22.0 ⁻¹²	27.4 ⁻¹⁵	32.1 ⁻¹⁸	39.1 ⁻¹⁹	53.9 ⁻²¹	74.9 ⁻²⁵	105.2 ⁻³⁰
.3	14.3 ⁻⁵	15.7 ⁻⁶	17.1 ⁻⁶	21.0 ⁻¹⁰	26.1 ⁻¹³	30.7 ⁻¹⁴	37.6 ⁻¹⁵	52.2 ⁻¹⁷	72.9 ⁻²⁰	102.7 ⁻²⁵
.4	13.9 ⁻⁴	15.2 ⁻⁵	16.5 ⁻⁶	20.2 ⁻⁸	25.0 ⁻¹¹	29.5 ⁻¹²	36.3 ⁻¹³	50.8 ⁻¹⁴	71.2 ⁻¹⁷	100.6 ⁻²¹
1.5	13.6	14.8	16.0	19.5	24.2	28.5	35.3	49.6	69.8	98.8
.6	13.3	14.4 ⁻⁴	15.6 ⁻⁴	19.0 ⁻⁵	23.4 ⁻⁸	27.7 ⁻⁸	34.5 ⁻⁸	48.6 ⁻¹⁰	68.6 ⁻¹²	97.3 ⁻¹⁵
.7	13.0	14.1	15.2 ⁻⁴	18.5	22.8 ⁻⁶	27.0 ⁻⁷	33.7 ⁻⁸	47.7 ⁻⁹	67.6 ⁻¹⁰	95.9 ⁻¹⁴
.8	12.8	13.8	14.9	18.1	22.3 ⁻⁵	26.4 ⁻⁶	33.0 ⁻⁷	47.0 ⁻⁷	66.7 ⁻⁹	94.7 ⁻¹²
.9	12.6	13.6	14.6	17.7	21.8 ⁻⁵	25.9 ⁻⁵	32.5 ⁻⁵	46.4 ⁻⁶	65.9 ⁻⁸	93.6 ⁻¹¹
2.0	12.4	13.3	14.4	17.4	21.4	25.5	32.0	45.8	65.2	92.7
.1	12.3	13.1	14.1	17.1	21.1	25.1	31.6	45.3 ⁻⁵	64.6 ⁻⁶	91.8 ⁻⁹
.2	12.1	13.0	13.9	16.9	20.8	24.7	31.2	44.9	64.0 ⁻⁶	91.0 ⁻⁸
.3	12.0	12.8	13.7	16.6	20.5	24.4	30.8	44.5	63.5 ⁻⁵	90.3 ⁻⁷
.4	11.9	12.7	13.6	16.4	20.2	24.2	30.5	44.1	63.1 ⁻⁴	89.6 ⁻⁶
2.5	11.8	12.6	13.4	16.3	20.0	23.9	30.3	43.8	62.7	89.0
.6	11.7	12.4	13.3	16.1	19.8	23.7	30.0	43.5	62.3	88.5
.7	11.6	12.4	13.2	15.9	19.6	23.5	29.8	43.2	61.9	88.0
.8	11.6	12.3	13.1	15.8	19.5	23.3	29.6	43.0	61.6	87.6
.9	11.5	12.2	13.0	15.7	19.4	23.1	29.4	42.8	61.3	87.2
3.0	11.4	12.1	12.9	15.6	19.2	23.0	29.3	42.6	61.0	86.8
.1	11.4	12.0	12.8	15.5	19.1	22.9	29.1	42.4	60.8	86.5
.2	11.3	12.0	12.7	15.4	19.0	22.8	29.0	42.2	60.6	86.2
.3	11.3	11.9	12.7	15.3	18.9	22.6	28.8	42.0	60.4	85.9
.4	11.2	11.9	12.6	15.2	18.8	22.5	28.7	41.9	60.2	85.6
3.5	11.2	11.8	12.6	15.1	18.7	22.4	28.6	41.8	60.0	85.4
.6	11.2	11.8	12.5	15.1	18.6	22.4	28.5	41.7	59.8	85.1
.7	11.1	11.7	12.4	15.0	18.6	22.3	28.4	41.5	59.6	84.9
.8	11.1	11.7	12.4	14.9	18.5	22.2	28.3	41.4	59.5	84.7
.9	11.1	11.7	12.4	14.9	18.4	22.2	28.2	41.3	59.3	84.5
4.0	11.0	11.6	12.3	14.8	18.4	22.1	28.2	41.2	59.2	84.3
.1	11.0	11.6	12.3	14.8	18.3	22.0	28.1	41.1	59.1	84.2
.2	11.0	11.6	12.2	14.7	18.3	22.0	28.0	41.1	59.0	84.0
.3	11.0	11.6	12.2	14.7	18.2	21.9	27.9	41.0	58.8	83.9
.4	11.0	11.5	12.2	14.6	18.2	21.9	27.9	40.9	58.7	83.7
4.5	10.9	11.5	12.2	14.6	18.1	21.8	27.8	40.8	58.6	83.6

TABLE 1b—continued
 Values of b° : $N = 6$ —continued

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
4.5	10.9	11.5	12.2	14.6	18.1	21.8	27.8	40.8	58.6	83.6
.6	10.9	11.5	12.1	14.6	18.1	21.8	27.8	40.8	58.6	83.5
.7	10.9	11.4	12.1	14.5	18.0	21.7	27.7	40.7	58.5	83.4
.8	10.9	11.4	12.1	14.5	18.0	21.7	27.7	40.6	58.4	83.3
.9	10.9	11.4	12.1	14.5	18.0	21.7	27.7	40.6	58.3	83.2
5.0	10.8	11.4	12.0	14.5	17.9	21.6	27.6	40.5	58.2	83.1
.1	10.8	11.4	12.0	14.4	17.9	21.6	27.6	40.4	58.2	83.0
.2	10.8	11.4	12.0	14.4	17.9	21.6	27.5	40.4	58.1	82.9
.3	10.8	11.4	12.0	14.4	17.8	21.5	27.5	40.3	58.1	82.8
.4	10.8	11.4	12.0	14.4	17.8	21.5	27.5	40.3	58.0	82.7
5.5	10.8	11.3	12.0	14.3	17.8	21.5	27.4	40.2	57.9	82.7
.6	10.8	11.3	11.9	14.3	17.8	21.4	27.4	40.2	57.9	82.6
.7	10.8	11.3	11.9	14.3	17.7	21.4	27.4	40.2	57.8	82.5
.8	10.8	11.3	11.9	14.3	17.7	21.4	27.4	40.1	57.8	82.5
.9	10.7	11.3	11.9	14.3	17.7	21.4	27.3	40.1	57.7	82.4
6.0	10.7	11.3	11.9	14.2	17.7	21.4	27.3	40.0	57.7	82.4
6.5	10.7	11.2	11.8	14.2	17.6	21.3	27.2	39.9	57.5	82.1
7.0	10.7	11.2	11.8	14.1	17.6	21.2	27.1	39.7	57.3	81.9

TABLE 2a
Values of φ_c°

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	57.86	51.85	46.70	35.27	27.95	24.45	21.70	19.48	18.52	18.08
.1	60.26	54.47	49.41	37.90	30.27	26.57	23.64	21.26	20.23	19.75
.2	60.36	56.80	51.85	40.33	32.48	28.62	25.52	23.00	21.90	21.39
.3	64.20	58.86	54.06	42.60	34.59	30.59	27.35	24.69	23.54	23.00
.4	65.83	60.71	56.05	44.72	36.60	32.48	29.12	26.34	25.13	24.56
1.5	67.27	62.36	57.86	46.70	38.51	34.30	30.83	27.95	26.68	26.09
.6	68.56	63.85	59.50	48.54	40.33	36.04	32.48	29.51	28.20	27.58
.7	69.72	65.20	61.00	50.25	42.05	37.71	34.07	31.02	29.67	29.03
.8	70.76	66.43	62.36	51.85	43.68	39.30	35.61	32.48	31.10	30.44
.9	71.70	67.44	63.62	53.36	45.23	40.83	37.09	33.90	32.48	31.81
2.0	72.56	68.46	64.77	54.75	46.70	42.29	38.51	35.27	33.83	33.14
.1	73.34	69.49	65.83	56.05	48.09	43.68	39.88	36.60	35.13	34.43
.2	74.06	70.35	66.81	57.27	49.41	45.00	41.20	37.89	36.40	35.69
.3	74.72	71.15	67.72	58.42	50.66	46.28	42.46	39.13	37.62	36.90
.4	75.33	71.88	68.56	59.50	51.85	47.50	43.68	40.33	38.80	38.08
2.5	75.89	72.56	69.34	60.51	52.98	48.66	44.85	41.48	39.95	39.22
.6	76.41	73.19	70.07	61.46	54.06	49.77	45.97	42.60	41.06	40.33
.7	76.90	73.78	70.76	62.36	55.08	50.84	47.05	43.68	42.13	41.40
.8	77.35	74.33	71.40	63.21	56.05	51.85	48.09	44.72	43.17	42.43
.9	77.77	74.85	72.00	64.02	56.98	52.83	49.09	45.73	44.18	43.43
3.0	78.17	75.33	72.56	64.77	57.86	53.77	50.05	46.70	45.15	44.40
.1	78.54	75.78	73.09	65.49	58.70	54.65	50.97	47.63	46.09	45.34
.2	78.89	76.21	73.59	66.17	59.50	55.50	51.85	48.54	47.00	46.25
.3	79.22	76.61	74.06	66.81	60.26	56.31	52.71	49.41	47.87	47.13
.4	79.53	76.99	74.51	67.42	61.00	57.10	53.53	50.25	48.72	47.98
3.5	79.82	77.35	74.93	68.00	61.70	57.86	54.32	51.07	49.54	48.81
.6	80.10	77.69	75.33	68.56	62.36	58.59	55.08	51.85	50.34	49.61
.7	80.36	78.02	75.71	69.09	63.00	59.28	55.81	52.61	51.11	50.38
.8	80.61	78.32	76.07	69.59	63.62	59.94	56.52	53.35	51.85	51.13
.9	80.85	78.61	76.41	70.07	64.21	60.58	57.20	54.06	52.57	51.85
4.0	81.07	78.89	76.74	70.53	64.77	61.20	57.86	54.74	53.27	52.56
.1	81.29	79.16	77.05	70.98	65.31	61.79	58.49	55.41	53.95	53.24
.2	81.49	79.41	77.35	71.40	65.83	62.36	59.10	56.05	54.60	53.90
.3	81.69	79.65	77.64	71.80	66.33	62.91	59.69	56.67	55.24	54.54
.4	81.87	79.88	77.91	72.19	66.81	63.44	60.27	57.28	55.85	55.16
4.5	82.05	80.10	78.17	72.56	67.27	63.96	60.82	57.86	56.45	55.76

TABLE 2a—continued
 Values of φ_0° —continued

$J \backslash r_0$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
4.5	82.05	80.10	78.17	72.56	67.27	63.96	60.82	57.86	56.45	55.76
.6	82.22 17	80.31 21	78.42 25	72.92 36	67.72 45	64.45 49	61.35 53	58.42 56	57.02 57	56.34 58
.7	82.39 17	80.51 20	78.66 24	73.26 34	68.15 43	64.93 48	61.86 51	58.97 55	57.58 56	56.91 57
.8	82.54 15	80.71 20	78.89 23	73.59 33	68.56 41	65.39 46	62.36 50	59.50 53	58.13 55	57.46 55
.9	82.69 15	80.89 18	79.11 22	73.91 30	68.96 38	65.83 44	62.85 49	60.01 51	58.65 52	57.99 53
5.0	82.84	81.07	79.33	74.21	69.34	66.26	63.31	60.51	59.17	58.51
.1	82.98 14	81.25 18	79.53 20	74.51 30	69.72 38	66.67 41	63.77 46	61.00 49	59.66 49	59.01 50
.2	83.11 13	81.41 16	79.73 20	74.79 28	70.08 36	67.08 41	64.20 43	61.47 47	60.15 49	59.50 49
.3	83.24 13	81.57 16	79.92 19	75.07 28	70.42 34	67.47 39	64.63 43	61.92 45	60.61 46	59.97 47
.4	83.36 12	81.72 15	80.10 18	75.33 26	70.76 32	67.84 37	65.04 40	62.36 43	61.07 44	60.44 44
5.5	83.48	81.87	80.28	75.59	71.08	68.21	65.44	62.79	61.51	60.88
.6	83.59 11	82.02 15	80.45 17	75.83 24	71.40 32	68.56 35	65.83 39	63.21 42	61.94 43	61.32 44
.7	83.71 12	82.15 13	80.61 16	76.07 24	71.70 30	68.90 34	66.21 38	63.62 41	62.36 42	61.75 43
.8	83.82 11	82.29 14	80.77 16	76.30 23	72.00 30	69.24 34	66.57 36	64.01 39	62.77 41	62.16 41
.9	83.92 10	82.42 12	80.92 15	76.53 21	72.28 28	69.56 31	66.93 34	64.39 38	63.17 38	62.56 39
6.0	84.02	82.54	81.07	76.74	72.56	69.87	67.27	64.77	63.55	62.95
6.5	84.48	83.11	81.75	77.73	73.83	71.31	68.86	66.49	65.34	64.77
7.0	84.87	83.60	82.33	78.58	74.93	72.56	70.25	68.01	66.91	66.37

TABLE 2b
Values of r_c sec φ_0 .

$J \backslash r_0$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975	1.0
1.0	0.376	0.405	0.437	0.551	0.679	0.769	0.861	0.955	1.002	1.026	1.049
.1	.403	.430	.461	.570	.695	.783	.873	.966	1.013	1.036	1.059
.2	.431	.457	.486	.590	.711	.797	.886	.978	1.024	1.047	1.070
.3	.460	.483	.511	.611	.729	.813	.901	.991	1.036	1.059	1.082
.4	.488	.511	.537	.633	.747	.830	.916	1.004	1.049	1.072	1.095
1.5	0.518	0.539	0.564	0.656	0.767	0.847	0.932	1.019	1.063	1.086	1.108
.6	.547	.567	.591	.680	.787	.866	.948	1.034	1.078	1.100	1.122
.7	.577	.596	.619	.704	.808	.885	.966	1.050	1.093	1.115	1.137
.8	.607	.625	.647	.729	.830	.905	.984	1.067	1.109	1.131	1.153
.9	.637	.654	.675	.754	.852	.925	1.003	1.084	1.126	1.147	1.169
2.0	0.667	0.684	0.704	0.780	0.875	0.946	1.022	1.102	1.144	1.164	1.185
.1	.698	.714	.733	.806	.898	.968	1.042	1.121	1.162	1.182	1.203
.2	.728	.744	.762	.832	.922	.990	1.063	1.140	1.180	1.200	1.221
.3	.759	.774	.791	.859	.947	1.013	1.084	1.160	1.199	1.219	1.239
.4	.790	.804	.821	.887	.971	1.036	1.106	1.180	1.219	1.239	1.258
2.5	0.821	0.834	0.850	0.914	0.997	1.060	1.128	1.201	1.239	1.259	1.278
.6	.851	.865	.880	.942	1.022	1.084	1.151	1.223	1.260	1.279	1.298
.7	.882	.895	.910	.970	1.048	1.108	1.174	1.244	1.281	1.300	1.319
.8	.913	.926	.940	.998	1.074	1.133	1.198	1.267	1.303	1.321	1.340
.9	.945	.956	.971	1.027	1.101	1.158	1.222	1.289	1.325	1.343	1.361
3.0	0.976	0.987	1.001	1.056	1.128	1.184	1.246	1.312	1.347	1.365	1.383
.1	1.007	1.018	1.031	1.085	1.155	1.210	1.270	1.336	1.370	1.387	1.405
.2	1.038	1.049	1.062	1.114	1.182	1.236	1.295	1.359	1.393	1.410	1.427
.3	1.069	1.080	1.092	1.143	1.210	1.262	1.320	1.383	1.416	1.433	1.450
.4	1.101	1.111	1.123	1.172	1.237	1.289	1.346	1.408	1.440	1.457	1.473
3.5	1.132	1.142	1.154	1.201	1.265	1.316	1.372	1.432	1.464	1.481	1.497
.6	1.163	1.173	1.184	1.231	1.293	1.343	1.398	1.457	1.488	1.505	1.521
.7	1.195	1.204	1.215	1.261	1.322	1.370	1.424	1.482	1.513	1.529	1.545
.8	1.226	1.235	1.246	1.291	1.350	1.397	1.450	1.508	1.538	1.554	1.569
.9	1.257	1.266	1.277	1.320	1.379	1.425	1.477	1.533	1.563	1.579	1.594
4.0	1.289	1.298	1.308	1.350	1.408	1.453	1.504	1.559	1.589	1.604	1.619
.1	1.320	1.329	1.339	1.380	1.437	1.481	1.531	1.585	1.614	1.629	1.644
.2	1.352	1.360	1.370	1.411	1.465	1.509	1.558	1.612	1.640	1.655	1.669
.3	1.383	1.391	1.401	1.441	1.494	1.537	1.585	1.638	1.666	1.681	1.695
.4	1.415	1.423	1.432	1.471	1.524	1.566	1.613	1.665	1.692	1.707	1.721
4.5	1.446	1.454	1.463	1.501	1.553	1.594	1.641	1.692	1.719	1.733	1.747

TABLE 2b—continued
 Values of r_c sec φ_o —continued

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975	1.0
4.5	1.446 ₃₂	1.454 ₃₁	1.463 ₃₂	1.501 ₃₁	1.553 ₂₉	1.594 ₂₉	1.64 ₂₈	1.692 ₂₇	1.719 ₂₆	1.733 ₂₆	1.747 ₂₆
.6	1.478 ₃₁	1.485 ₃₂	1.495 ₃₁	1.532 ₃₀	1.582 ₃₀	1.623 ₂₉	1.669 ₂₈	1.719 ₂₇	1.745 ₂₇	1.759 ₂₇	1.773 ₂₆
.7	1.509 ₃₂	1.517 ₃₁	1.526 ₃₁	1.562 ₃₁	1.612 ₂₉	1.652 ₂₉	1.697 ₂₈	1.746 ₂₇	1.772 ₂₇	1.786 ₂₇	1.799 ₂₇
.8	1.541 ₃₂	1.548 ₃₂	1.557 ₃₁	1.593 ₃₀	1.641 ₃₀	1.681 ₂₉	1.725 ₂₈	1.773 ₂₈	1.799 ₂₇	1.813 ₂₆	1.826 ₂₇
.9	1.573 ₃₁	1.580 ₃₁	1.588 ₃₂	1.623 ₃₁	1.671 ₃₀	1.710 ₂₉	1.753 ₂₈	1.801 ₂₇	1.826 ₂₇	1.839 ₂₇	1.853 ₂₇
5.0	1.604 ₃₂	1.611 ₃₂	1.620 ₃₁	1.654 ₃₁	1.701 ₃₀	1.739 ₂₉	1.781 ₂₉	1.828 ₂₈	1.853 ₂₈	1.866 ₂₈	1.880 ₂₇
.1	1.636 ₃₁	1.643 ₃₁	1.651 ₃₁	1.685 ₃₀	1.731 ₃₀	1.768 ₂₉	1.810 ₂₈	1.856 ₂₈	1.881 ₂₈	1.894 ₂₇	1.907 ₂₇
.2	1.667 ₃₂	1.674 ₃₁	1.682 ₃₁	1.715 ₃₁	1.761 ₃₀	1.797 ₂₉	1.838 ₂₉	1.884 ₂₈	1.909 ₂₇	1.921 ₂₇	1.934 ₂₇
.3	1.699 ₃₁	1.705 ₃₂	1.713 ₃₂	1.746 ₃₁	1.791 ₃₀	1.826 ₃₀	1.867 ₂₉	1.912 ₂₈	1.936 ₂₈	1.948 ₂₈	1.961 ₂₈
.4	1.730 ₃₂	1.737 ₃₁	1.745 ₃₁	1.777 ₃₁	1.821 ₃₀	1.856 ₂₉	1.896 ₂₉	1.940 ₂₈	1.964 ₂₈	1.976 ₂₈	1.989 ₂₇
5.5	1.762 ₃₂	1.768 ₃₂	1.776 ₃₂	1.808 ₃₁	1.851 ₃₀	1.885 ₃₀	1.925 ₂₉	1.968 ₂₉	1.992 ₂₈	2.004 ₂₈	2.016 ₂₈
.6	1.74 ₃₂	1.800 ₃₂	1.808 ₃₁	1.839 ₃₀	1.881 ₃₀	1.915 ₃₀	1.954 ₂₉	1.997 ₂₈	2.020 ₂₈	2.032 ₂₈	2.044 ₂₈
.7	1.826 ₃₁	1.832 ₃₁	1.839 ₃₁	1.869 ₃₁	1.911 ₃₀	1.945 ₂₉	1.983 ₂₉	2.025 ₂₉	2.048 ₂₈	2.060 ₂₈	2.072 ₂₈
.8	1.857 ₃₂	1.863 ₃₂	1.870 ₃₂	1.900 ₃₁	1.941 ₃₁	1.974 ₃₀	2.012 ₂₉	2.054 ₂₉	2.076 ₂₉	2.088 ₂₈	2.100 ₂₈
.9	1.889 ₃₁	1.895 ₃₁	1.902 ₃₁	1.931 ₃₁	1.972 ₃₀	2.004 ₃₀	2.041 ₃₀	2.083 ₂₈	2.105 ₂₈	2.116 ₂₈	2.128 ₂₈
6.0	1.920	1.926	1.933	1.962	2.002	2.034	2.071	2.111	2.133	2.144	2.156
6.5	2.079	2.084	2.091	2.117	2.154	2.184	2.218	2.256	2.277	2.287	2.298
7.0	2.237	2.242	2.248	2.273	2.307	2.335	2.367	2.403	2.422	2.432	2.442

TABLE 2c Values of τ

$J \backslash r_c$	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975	Δ
1.0	0.742	1.178	1.788	2.292	2.873	3.532	3.891	4.077	
.1	0.824	1.260	1.870	2.374	2.956	3.615	3.973	4.160	82
.2	0.914	1.350	1.961	2.465	3.046	3.705	4.063	4.250	90
.3	1.012	1.449	2.059	2.563	3.144	3.803	4.162	4.348	98
.4	1.118	1.555	2.165	2.669	3.250	3.909	4.268	4.454	106
									114
1.5	1.232	1.668	2.279	2.783	3.364	4.023	4.382	4.568	
.6	1.354	1.790	2.401	2.904	3.486	4.145	4.503	4.690	122
.7	1.484	1.920	2.530	3.034	3.615	4.274	4.633	4.819	130
.8	1.621	2.057	2.668	3.172	3.753	4.412	4.770	4.957	138
.9	1.767	2.203	2.813	3.317	3.898	4.557	4.916	5.102	145
									153
2.0	1.920	2.356	2.966	3.470	4.051	4.710	5.069	5.255	
.1	2.081	2.517	3.127	3.631	4.212	4.871	5.230	5.416	161
.2	2.250	2.686	3.296	3.800	4.381	5.040	5.399	5.585	169
.3	2.426	2.862	3.473	3.977	4.558	5.217	5.575	5.762	177
.4	2.611	3.047	3.657	4.161	4.743	5.401	5.760	5.946	185
									192
2.5	2.803	3.239	3.850	4.354	4.935	5.594	5.952	6.139	
.6	3.004	3.440	4.050	4.554	5.135	5.794	6.153	6.339	200
.7	3.212	3.648	4.258	4.762	5.343	6.002	6.361	6.547	208
.8	3.428	3.864	4.474	4.978	5.559	6.218	6.577	6.763	216
.9	3.651	4.088	4.698	5.202	5.783	6.442	6.801	6.987	224
									232
3.0	3.883	4.319	4.930	5.433	6.015	6.674	7.032	7.219	
.1	4.123	4.559	5.169	5.673	6.254	6.913	7.272	7.458	240
.2	4.370	4.806	5.417	5.920	6.502	7.161	7.519	7.706	247
.3	4.625	5.061	5.672	6.176	6.757	7.416	7.774	7.961	255
.4	4.888	5.324	5.935	6.439	7.020	7.679	8.038	8.224	263
									271
3.5	5.159	5.595	6.206	6.710	7.291	7.950	8.309	8.495	
.6	5.438	5.874	6.485	6.989	7.570	8.229	8.587	8.774	279
.7	5.725	6.161	6.771	7.275	7.857	8.516	8.874	9.061	287
.8	6.019	6.455	7.066	7.570	8.151	8.810	9.169	9.355	295
.9	6.322	6.758	7.368	7.872	8.454	9.112	9.471	9.657	302
									310
4.0	6.632	7.068	7.678	8.182	8.764	9.423	9.781	9.968	
.1	6.950	7.386	7.997	8.500	9.082	9.741	10.099	10.286	318
.2	7.276	7.712	8.322	8.826	9.408	10.067	10.425	10.612	326
.3	7.610	8.046	8.656	9.160	9.741	10.400	10.759	10.945	334
.4	7.951	8.387	8.998	9.502	10.083	10.742	11.100	11.287	342
									350
4.5	8.301	8.737	9.347	9.851	10.433	11.092	11.450	11.637	
.6	8.658	9.094	9.705	10.209	10.790	11.449	11.807	11.994	357
.7	9.024	9.460	10.070	10.574	11.155	11.814	12.173	12.359	365
.8	9.397	9.833	10.443	10.947	11.528	12.187	12.546	12.732	373
.9	9.778	10.214	10.824	11.328	11.909	12.568	12.927	13.113	381
									389
5.0	10.166	10.602	11.213	11.717	12.298	12.957	13.315	13.502	
.1	10.563	10.999	11.609	12.113	12.695	13.353	13.712	13.899	397
.2	10.967	11.403	12.014	12.518	13.099	13.758	14.116	14.303	404
.3	11.380	11.816	12.426	12.930	13.511	14.170	14.529	14.715	412
.4	11.800	12.236	12.846	13.350	13.932	14.590	14.949	15.135	420
									428
5.5	12.228	12.664	13.274	13.778	14.360	15.018	15.377	15.564	
.6	12.664	13.100	13.710	14.214	14.796	15.454	15.813	16.000	436
.7	13.108	13.544	14.154	14.658	15.239	15.898	16.257	16.444	444
.8	13.559	13.995	14.606	15.110	15.691	16.350	16.708	16.895	452
.9	14.019	14.455	15.065	15.569	16.150	16.809	17.168	17.355	460
									467
6.0	14.486	14.922	15.532	16.036	16.618	17.276	17.635	17.822	
6.5	16.940	17.376	17.987	18.491	19.072	19.731	20.090	20.276	
7.0	19.591	20.027	20.638	21.141	21.723	22.382	22.740	22.927	

TABLE 2d Values of ζ

$J \backslash r_0$	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	0.111	0.265	0.536	0.802	1.149	1.589	1.848	1.988
.1	.124	.283	.561	.831	1.182	1.627	1.887	2.028
.2	.137	.304	.588	.863	1.218	1.667	1.930	2.072
.3	.152	.326	.618	.897	1.258	1.711	1.977	2.120
.4	.168	.350	.649	.934	1.300	1.759	2.027	2.171
1.5	0.185	0.375	0.684	0.974	1.346	1.810	2.081	2.227
.6	.203	.403	.720	1.017	1.394	1.865	2.139	2.286
.7	.223	.432	.759	1.062	1.446	1.923	2.201	2.349
.8	.243	.463	.800	1.110	1.501	1.985	2.266	2.416
.9	.265	.496	.844	1.161	1.559	2.051	2.335	2.487
2.0	0.288	0.530	0.890	1.214	1.621	2.120	2.408	2.562
.1	.312	.566	.938	1.271	1.685	2.192	2.484	2.640
.2	.337	.604	.989	1.330	1.752	2.268	2.564	2.723
.3	.364	.644	1.042	1.392	1.823	2.348	2.648	2.809
.4	.392	.686	1.097	1.456	1.897	2.431	2.736	2.899
2.5	0.420	0.729	1.155	1.524	1.974	2.517	2.827	2.993
.6	.451	.774	1.215	1.594	2.054	2.607	2.922	3.090
.7	.482	.821	1.277	1.667	2.137	2.701	3.021	3.192
.8	.514	.869	1.342	1.742	2.224	2.798	3.124	3.297
.9	.548	.920	1.409	1.821	2.313	2.899	3.230	3.406
3.0	0.582	0.972	1.479	1.902	2.406	3.003	3.340	3.519
.1	.618	1.026	1.551	1.986	2.502	3.111	3.454	3.636
.2	.656	1.081	1.625	2.072	2.601	3.222	3.572	3.757
.3	.694	1.139	1.702	2.161	2.703	3.337	3.693	3.881
.4	.733	1.198	1.780	2.254	2.808	3.456	3.818	4.009
3.5	0.774	1.259	1.862	2.348	2.916	3.577	3.947	4.141
.6	.816	1.322	1.945	2.446	3.028	3.703	4.079	4.277
.7	.859	1.386	2.031	2.546	3.143	3.832	4.215	4.417
.8	.903	1.452	2.120	2.649	3.260	3.964	4.355	4.561
.9	.948	1.520	2.210	2.755	3.381	4.101	4.499	4.708
4.0	0.995	1.590	2.304	2.864	3.505	4.240	4.646	4.859
.1	1.043	1.662	2.399	2.975	3.633	4.383	4.797	5.014
.2	1.091	1.735	2.497	3.089	3.763	4.530	4.952	5.173
.3	1.141	1.810	2.597	3.206	3.897	4.680	5.110	5.336
.4	1.193	1.887	2.699	3.326	4.033	4.834	5.273	5.502
4.5	1.245	1.966	2.804	3.448	4.173	4.991	5.439	5.673
.6	1.299	2.046	2.911	3.573	4.316	5.152	5.609	5.847
.7	1.354	2.128	3.021	3.701	4.462	5.316	5.782	6.025
.8	1.409	2.212	3.133	3.831	4.611	5.484	5.959	6.207
.9	1.467	2.298	3.247	3.965	4.764	5.656	6.140	6.393
5.0	1.525	2.386	3.364	4.101	4.919	5.831	6.325	6.582
.1	1.584	2.475	3.483	4.240	5.078	6.009	6.513	6.776
.2	1.645	2.566	3.604	4.381	5.240	6.191	6.705	6.973
.3	1.707	2.659	3.728	4.526	5.405	6.377	6.901	7.174
.4	1.770	2.753	3.854	4.673	5.573	6.566	7.101	7.379
5.5	1.834	2.849	3.982	4.822	5.744	6.758	7.304	7.587
.6	1.900	2.947	4.113	4.975	5.918	6.954	7.511	7.800
.7	1.966	3.047	4.246	5.130	6.096	7.154	7.722	8.016
.8	2.034	3.149	4.382	5.288	6.276	7.357	7.936	8.236
.9	2.103	3.252	4.520	5.449	6.460	7.564	8.155	8.460
6.0	2.173	3.357	4.660	5.613	6.647	7.774	8.377	8.688
6.5	2.541	3.910	5.396	6.472	7.629	8.879	9.543	9.885
7.0	2.939	4.506	6.191	7.399	8.689	10.072	10.802	11.177

TABLE 2e Values of q

These values are used as they stand in Eq. 33 but they must be divided by $180/\pi$ for use in Eq. 31 if β is expressed in degrees

$J \backslash r_c$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
1.0	0.103	0.128	0.162	0.325	0.607	0.881	1.237	1.686	1.949	2.091
.1	0.127 24	0.154 26	0.190 28	0.359 34	0.650 43	0.929 48	1.291 54	1.745 59	2.011 62	2.155 64
.2	0.155 28	0.184 30	0.222 32	0.399 40	0.697 47	0.983 54	1.350 59	1.811 66	2.080 69	2.225 70
.3	0.188 33	0.219 35	0.259 37	0.443 44	0.750 53	1.042 59	1.416 66	1.884 73	2.156 76	2.303 78
.4	0.226 38	0.259 40	0.300 41	0.492 49	0.809 59	1.107 65	1.488 72	1.963 79	2.239 83	2.387 84
	43	45	47	55	65	72	79	86	90	93
1.5	0.269	0.304	0.347	0.547	0.874	1.179	1.567	2.049	2.329	2.480
.6	0.317 48	0.354 50	0.400 53	0.608 61	0.945 71	1.257 78	1.653 86	2.143 94	2.427 98	2.579 99
.7	0.372 55	0.410 56	0.459 59	0.676 68	1.022 77	1.342 85	1.746 93	2.244 101	2.533 106	2.687 108
.8	0.433 61	0.473 63	0.524 65	0.749 73	1.107 85	1.434 92	1.846 100	2.353 109	2.646 113	2.803 116
.9	0.501 68	0.543 70	0.596 72	0.830 81	1.198 91	1.534 100	1.955 109	2.471 118	2.768 122	2.927 124
	75	77	79	88	99	108	116	125	130	133
2.0	0.576	0.620	0.675	0.918	1.297	1.642	2.071	2.596	2.898	3.060
.1	0.658 82	0.704 84	0.762 87	1.014 96	1.404 107	1.757 115	2.196 125	2.730 134	3.037 139	3.201 141
.2	0.749 91	0.797 93	0.857 95	1.118 104	1.520 116	1.881 124	2.329 133	2.874 144	3.186 149	3.352 151
.3	0.847 98	0.897 100	0.960 103	1.230 112	1.644 124	2.014 133	2.471 142	3.026 152	3.343 157	3.513 161
.4	0.954 107	1.006 109	1.071 111	1.351 121	1.776 132	2.156 142	2.623 152	3.188 162	3.511 168	3.683 170
	117	119	121	130	142	151	161	172	177	180
2.5	1.071	1.125	1.192	1.481	1.918	2.307	2.784	3.360	3.688	3.863
.6	1.196 125	1.252 127	1.322 130	1.620 139	2.070 152	2.468 161	2.955 171	3.542 182	3.876 188	4.053 190
.7	1.331 135	1.390 138	1.462 140	1.769 149	2.232 162	2.639 171	3.137 182	3.735 193	4.074 198	4.255 202
.8	1.477 146	1.537 147	1.612 150	1.929 160	2.403 171	2.821 182	3.329 192	3.938 203	4.284 210	4.467 212
.9	1.633 156	1.695 158	1.772 160	2.099 170	2.586 183	3.013 192	3.532 203	4.153 215	4.504 220	4.691 224
	167	169	171	181	194	204	214	226	232	235
3.0	1.800	1.864	1.943	2.280	2.780	3.217	3.746	4.379	4.736	4.926
.1	1.98 18	2.04 18	2.126 183	2.472 192	2.985 205	3.432 215	3.972 226	4.617 238	4.980 244	5.173 247
.2	2.17 19	2.24 20	2.320 194	2.676 204	3.202 217	3.659 227	4.210 238	4.866 249	5.236 256	5.433 260
.3	2.37 20	2.44 20	2.526 206	2.892 216	3.431 229	3.898 239	4.461 251	5.129 263	5.505 269	5.705 272
.4	2.58 21	2.66 22	2.745 219	3.120 228	3.672 241	4.150 252	4.724 263	5.404 275	5.787 282	5.990 285
	23	22	231	242	254	264	276	289	295	298
3.5	2.81	2.88	2.976	3.362	3.926	4.414	5.000	5.693	6.082	6.288
.6	3.05 24	3.13 25	3.221 245	3.616 254	4.194 268	4.692 278	5.290 290	5.995 302	6.391 309	6.600 312
.7	3.30 25	3.38 25	3.479 258	3.884 268	4.475 281	4.984 292	5.593 303	6.311 316	6.714 323	6.926 326
.8	3.57 27	3.65 27	3.751 272	4.166 282	4.770 295	5.289 305	5.910 317	6.641 330	7.051 337	7.267 341
.9	3.85 28	3.93 28	4.037 286	4.462 296	5.080 310	5.610 321	6.242 332	6.986 345	7.402 351	7.622 355
	30	30	301	310	324	334	347	360	367	372
4.0	4.15	4.23	4.338	4.772	5.404	5.944	6.589	7.346	7.769	7.994
.1	4.46 31	4.55 32	4.653 315	5.098 326	5.744 340	6.294 350	6.951 362	7.721 375	8.151 382	8.378 384
.2	4.79 33	4.88 33	4.985 332	5.439 341	6.098 354	6.659 365	7.329 378	8.112 391	8.549 398	8.779 401
.3	5.13 34	5.22 34	5.331 346	5.796 357	6.468 370	7.041 382	7.722 393	8.518 406	8.963 414	9.197 418
.4	5.49 36	5.58 36	5.695 364	6.169 373	6.855 387	7.438 397	8.132 410	8.942 424	9.393 430	9.631 434
	37	38	379	390	403	414	426	440	447	451
4.5	5.86	5.96	6.074	6.559	7.258	7.852	8.558	9.382	9.840	10.082

TABLE 2e—continued Values of q —continued

These values are used as they stand in Eq. 33 but they must be divided by $180/\pi$ for use in Eq. 31 if β is expressed in degrees

$J \backslash r_e$	0.2	0.25	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
4.5	5.86	5.96	6.074	6.559	7.258	7.852	8.558	9.382	9.840	10.082
.6	6.25 39	6.35 39	6.470 396	6.965 406	7.678 420	8.284 432	9.002 444	9.839 457	10.304 464	10.550 468
.7	6.66 41	6.76 41	6.884 414	7.389 424	8.116 438	8.732 448	9.462 460	10.313 474	10.786 482	11.035 485
.8	7.09 43	7.19 43	7.315 431	7.830 441	8.571 455	9.199 467	9.941 479	10.805 492	11.285 499	11.539 504
.8	7.09 45	7.19 45	7.315 450	7.830 460	8.571 473	9.199 484	9.941 497	10.805 511	11.285 519	11.539 521
.9	7.54 46	7.64 46	7.765 467	8.290 478	9.044 492	9.683 503	10.438 515	11.316 529	11.804 536	12.060 541
5.0	8.00	8.10	8.232	8.768	9.536	10.186	10.953	11.845	12.340	12.601
.1	8.47 47	8.59 49	8.719 487	9.264 496	10.047 511	10.707 521	11.487 534	12.393 548	12.895 555	13.160 559
.2	8.98 51	9.09 50	9.225 506	9.780 516	10.576 529	11.248 541	12.041 554	12.961 568	13.471 576	13.738 678
.3	8.98 52	9.09 52	9.225 525	9.780 536	10.576 549	11.248 560	12.041 573	12.961 586	13.471 594	13.738 598
.3	9.50 54	9.61 55	9.750 544	10.316 554	11.125 568	11.808 580	12.614 592	13.547 607	14.065 614	14.336 618
.4	10.04 56	10.16 56	10.294 566	10.870 576	11.693 590	12.388 601	13.206 614	14.154 628	14.679 635	14.954 640
5.5	10.60	10.72	10.860	11.446	12.283	12.989	13.820	14.782	15.314	15.594
.6	11.18 58	11.30 58	11.446 586	12.042 596	12.893 610	13.610 621	14.454 634	15.430 648	15.970 656	16.254 660
.7	11.79 61	11.91 61	12.052 606	12.659 617	13.524 631	14.253 643	15.109 655	16.099 669	16.647 677	16.935 681
.8	12.41 62	12.53 62	12.681 629	13.297 638	14.177 653	14.917 664	15.786 677	16.790 691	17.346 699	17.637 702
.8	12.41 64	12.53 65	12.681 649	13.297 660	14.177 674	14.917 685	15.786 698	16.790 713	17.346 720	17.637 724
.9	13.05 68	13.18 67	13.330 672	13.957 682	14.851 695	15.602 707	16.484 720	17.503 735	18.066 743	18.361 747
6.0	13.73	13.85	14.002	14.639	15.546	16.309	17.204	18.238	18.809	19.108
6.5	17.40	17.54	17.708	18.396	19.374	20.194	21.154	22.259	22.869	23.188
7.0	21.70	21.85	22.024	22.762	23.811	24.688	25.714	26.892	27.541	27.881

TABLE 3
Integrating coefficients

TABLE 3a
(Eight points, square root, $\alpha = \frac{1}{2}$, see R. & M. 2043⁹)

r_c	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975
r_c^2	0.09	0.2025	0.36	0.49	0.64	0.81	0.9025	0.951
Integrating coefficient	0.03307	0.16668	0.13147	0.14282	0.16079	0.14466	0.05481	0.06745

TABLE 3b
Integrating coefficients for root loss

Spinner Radius r_{c0}	Coefficient at 0.2 radius	Coefficient at 0.25 radius	Coefficient at 0.3 radius
0.10	0.06548	-0.00269	0.01721
0.11	0.05946	+0.00222	0.01622
0.12	0.05320	0.00719	0.01521
0.13	0.04678	0.01211	0.01421
0.14	0.04029	0.01686	0.01325
0.15	0.03385	0.02130	0.01235
0.16	0.02756	0.02529	0.01155
0.17	0.02155	0.02869	0.01086
0.18	0.01594	0.03134	0.01032
0.19	0.01087	0.03306	0.00997
0.20	0.00648	0.03367	0.00985
0.21	0.00290	0.03303	0.00997
0.22	+0.00018	0.03114	0.01028
0.23	-0.00169	0.02810	0.01069
0.24	-0.00275	0.02409	0.01106
0.25	-0.00308	0.01935	0.01123
0.26	-0.00280	0.01422	0.01098
0.27	-0.00209	0.00912	0.01007
0.28	-0.00118	0.00460	0.00819
0.29	-0.00037	0.00130	0.00497
0.30	0	0	0

TABLE 4

Specimen Calculations

The following specimen calculations refer to a 5-bladed airscrew operating at $J = 2.65$

r_c	0.95	0.95	0.95	
t/c	0.062	0.062	0.062	From blade details.
$r_c \sec \varphi_0$	1.270	1.270	1.270	Table 2b.
M	0.758	0.882	0.882	Equation (40).
θ°	45.0	45.0	49.0	Blade details.
ε_0°	2.94	2.94	2.94	Blade details.
φ_0°	41.59	41.59	41.59	Table 2a.
s	0.064	0.064	0.064	Blade details.
A_0	0.1	0.1	0.1	Table 1 of R. & M. 2036 ⁷ .
a_0	156.2	156.2	156.2	Equation (47).
M_L'	0.784	0.784	0.784	Table 3 of R. & M. 2036 ⁷ .
$(1 - M_L'^2)^{1/2}$	0.621	0.621	0.621	Table 2 of R. & M. 2036 ⁷ .
$(a')^\circ$	97.0	97.0	97.0	Equation (49).
b°	69.6	69.6	69.6	Table 1b.
$(\theta - \varphi_0 + \varepsilon_0)^\circ$	6.35	6.35	10.35	
$(a' + b)^\circ$	166.6	166.6	166.6	
$(\alpha_0')^\circ$	3.70	3.70	6.02	$a'(\theta - \varphi_0 + \varepsilon_0)/(a' + b)$.
M_L	0.782	0.782	0.750	Table 3 of R. & M. 2036 ⁷ . $M_L = M_L'$ when $\alpha_0' \leq 3^\circ$.
$M - M_L$	<0	0.100	0.132	
C_{LS}	—	-0.002	-0.043	Table 4 of R. & M. 2036 ⁷ .
$(1 - M^2)^{1/2}$	0.652	—	—	Table 2 of R. & M. 2036 ⁷ .
$(1 - M_L'^2)^{1/2}$	—	0.623	0.661	Table 2 of R. & M. 2036 ⁷ .
$(1 - M_L'^2)^{1/2} C_{LS}/A_0$	—	-0.01	-0.28	
ε°	2.94	2.93	2.66	§ 6.
a°	101.8	97.3	103.3	§ 6.
$(\theta - \varphi_0 + \varepsilon)^\circ$	6.35	6.34	10.07	
$(a + b)^\circ$	171.4	166.9	172.9	
sC_L	0.0371	0.0380	0.0583	Equation (13).
C_L	0.58	0.59	0.91	
$(asC_L)^\circ$	3.78	3.70	6.02	
α_0°	3.78	3.71	6.30	Equal to asC_L in Range 1 and $asC_L - (1 - M_L'^2)^{1/2} C_{LS}/A_0$ in Range 2.
β°	2.58	2.64	4.06	Equation (9).
φ°	44.17	44.23	45.65	Equation (7).
$(\alpha_0 + \varphi)^\circ$	47.95	47.94	51.95	Check = $\theta + \varepsilon_0$.
M_D	0.742	0.746	0.592	Table 8 of R. & M. 2036 ⁷ .
$M - M_D$	0.016	0.136	0.290	
B_0	0.998	0.997	1.124	Table 6 of R. & M. 2036 ⁷ .
C_0	0.00804	0.00804	0.00804	Table 7 of R. & M. 2036 ⁷ .
C_{D0}	0.0080	0.0080	0.0090	Equation (52).
C_{Ds}	0.0006	0.0342	0.0860	Table 9 of R. & M. 2036 ⁷ .
C_D	0.0086	0.0422	0.0950	Equation (51).
sC_D	0.00055	0.00270	0.00608	
$\cos \varphi$	0.717	0.717	0.699	
$\sin \varphi$	0.697	0.698	0.715	
$sC_L \sin \varphi$	0.0258	0.0265	0.0417	
$sC_D \cos \varphi$	0.0004	0.0019	0.0042	
Sum	0.0262	0.0284	0.0459	$(sC_L \sin \varphi + sC_D \cos \varphi)$.
ζ	2.972	2.972	2.972	Table 2d.
q_c	0.0778	0.0844	0.1365	Equation (18).
q	3.975	3.975	3.975	Table 2e.
\dot{p}_{o1}	0.0066	0.0070	0.0164	Equation (31).
\dot{p}_{o0}	0.0020	0.0020	0.0023	Equation (33).
\dot{p}_{os}	0.0002	0.0087	0.0219	Equation (33).

Specimen Column for Blade Root Loss

r_e	0.25	
t/c	0.23	Blade details.
θ°	74.60	Blade details.
φ_0°	73.48	Table 2a.
ε_0°	6.20	Blade details.
M	0.610	Equation (40) with Table 2b.
s	0.51	Blade details
a°	21.4	§ 6.
b°	12.0	Table 1b.
$(\theta - \varphi_0 + \varepsilon_0)^\circ$..	7.32	
$(a + b)^\circ$	33.4	
α_0°	4.68	Equation (56).
C_D	0.028	Fig. 12 of R. & M. 2036 ⁷ .
q	1.319	Table 2e.
$q s C_D$	0.0189	

Example of Integrations

r_e	0.3	0.45	0.6	0.7	0.8	0.9	0.95	0.975	Integrated results
q_c	0.1354	0.1740	0.1784	0.1561	0.1340	0.1030	0.0844	0.0662	$k_q = 0.1248$
\dot{p}_{c1}	0.0208	0.0236	0.0203	0.0150	0.0115	0.0081	0.0070	0.0058	$k_{p1} = 0.0132$
\dot{p}_{c0}	0.0058	0.0045	0.0038	0.0033	0.0029	0.0024	0.0020	0.0018	$k_{p0} = 0.0030$
\dot{p}_{os}	0	0	0.0012	0.0021	0.0049	0.0081	0.0087	0.0074	$k_{ps} = 0.0034$

Root Loss

r_e	0.20	0.25	0.30	Integral
$q s C_D$	0.1090	0.0189	0.0028	$\Delta k_P = 0.00137$

$\Delta \eta = 0.011$

$k_{P1}/k_q = 0.106$

$k_{P0}/k_q = 0.024$

$k_{PS}/k_q = 0.027$

$k_P/k_q = 0.157$

$\eta = 0.843$

Root loss $\Delta \eta = 0.011$

Final $\eta = 0.832$